

FLIGHT DATA FILE

Crew Activity Plan

STS-26

(Cycle 3R Trajectory)

Mission Operations Directorate
Operations Division

Final, Rev C
September 16, 1988

<u>EVENT</u>	<u>CDT</u>		
Launch	09:03	Tue	Oct 4
TDRS Deploy	15:17	Tue	Oct 4
EDW Landing	10:00	Sat	Oct 8



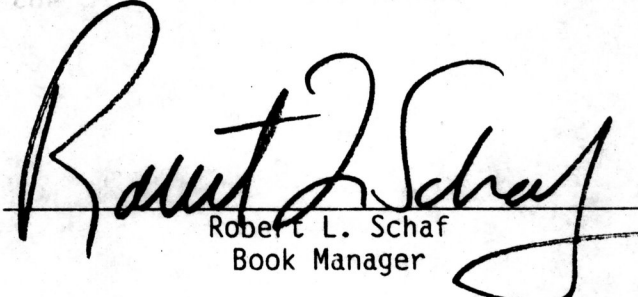
National Aeronautics and
Space Administration

Lyndon B. Johnson Space Center
Houston, Texas

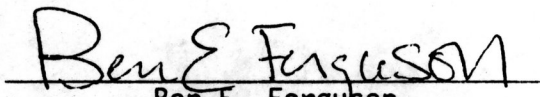
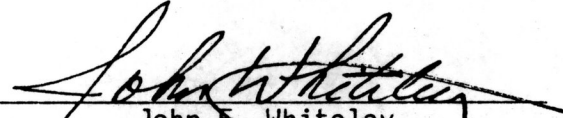
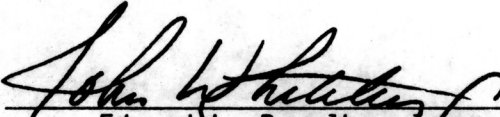
MISSION OPERATIONS DIRECTORATE

CREW ACTIVITY PLAN
STS-26FINAL, REVISION C
September 16, 1988

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482#: CAP-566 CAP-567 CAP-568	EXEMPTION#: 1

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Chief, Operations Division

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STS-26 CREW ACTIVITY PLAN

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(*) - Omit from flight book

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INTRODUCTION

The STS-26 Crew Activity Plan is the flight data file article that contains the on-orbit timeline. It does not contain the detailed timelines that are covered in the Ascent, Post Insertion, Deorbit Prep, and Entry Checklists. The detailed deploy procedures required during the deploy timeframe are in the IUS Deploy Checklist. For continuity, the entire flight is shown in the summary level timeline.

This on-orbit timeline satisfies the objectives specified in the Flight Definition and Requirements Directive and the requirements of the STS-26 Flight Requirements Document. This Crew Activity Plan, as part of the Flight Data File, meets the specifications of the Cycle 3 version of the STS-26B Flight Initialization Data Pack. (PCN-3)

The flight profile used for this Crew Activity Plan is for a launch date of October 4, 1988, at 09:03 CDT. Orbiter inertial attitudes are expressed relative to a Mean of 1950 (M50) reference frame.

JSC-22354 Crew Scheduling Constraints (Appendix K, CPMP), Basic, March 1987, has been used to build the timelines contained in this Crew Activity Plan. The Flight Day 2 unscheduled EVA plan in Section 4-D is specifically addressed in approved Exemption Request #1 to this document.

The timelines published in this Crew Activity Plan assume that crew wakeup on launch day will occur at L-4:30, increasing the crew awake time before launch to 4 hours and 50 minutes.

INTRODUCTION

The STS-51-L Crew Activity Plan is the flight data file article that contains the on-orbit timeline. It does not contain the detailed timeline that are covered in the Ascent, Post Insertion, Descent, and Entry Checklists. The detailed deployment procedures required during the deployment are in the US Deploy Checklist. For continuity, the entire flight is shown in the summary level timeline.

This on-orbit timeline satisfies the objectives specified in the Definition and Requirements Directive and the requirements of the STS-51-L Flight Requirements Document. This Crew Activity Plan is part of the STS-51-L Flight Data File, meets the specifications of the Cycle 3 version of STS-51-L Flight Initialization Data Pack. (FCH-3)

The flight profile used for this Crew Activity Plan is for a launch date of October 4, 1988, at 09:03 GMT. Other initial conditions are expressed relative to a Mean of 1950 (MEO) reference frame.

USC-23354 Crew Scheduling Constraints (Appendix K, CPMF, Basic Mission 1987) has been used to build the timeline contained in this Crew Activity Plan. The Flight Day 2 unscheduled EVA plan in Section 4-B is specifically addressed in approved Exemption Request #1 to this document.

The timeline published in this Crew Activity Plan assumes that crew wakeup on launch day will occur at 06:00 GMT. Crew awake time before launch to 4 hours and 30 minutes.

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STS-26 OVERVIEW

[illegible]

FLIGHT		STS 26	EDITION	FINAL. REV C	PUB. DATE 9/16/88																				
MET :003	16	17	18	19	20	21	22	23	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16
FD 5	ADSF 4	POST SLEEP	IMU S	POST SLEEP	DEORBIT PREP																				
DAY/NIGHT																									
ORBIT	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75								
NOTES:																									

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SUMMARY TIMELINE

[illegible]

GMT (D:H:M)	MET (D:H:M)	CDT (D:H:M)	FD/DOY	BETA	MOON	HOUSTON DATE	FLIGHT	EDITION	PUB. DATE
279:02:03 / 279:14:03	000:12:00 / 001:00:00	278:21:03 / 279:09:03	02 / 278 CDT	-22.7		OCTOBER 4, 1988	STS 26	FINAL, REV C	9/16/88
CDT :278									
FD :01									
MET :000									
STS	SLEEP								
PL									
DAY/NIGHT									
ORBIT									
GSTDN									
COVERAGE									
TDRS									
ATTITUDE									
NOTES:									

[illegible]

GMT (D:H:M)	MET	(D:H:M)	CDT	(D:H:M)	FD/DOY	BETA	MOON	HOUSTON DATE	FLIGHT	EDITION	PUB. DATE
280:14:03 / 281:02:03	002:00:00 / 002:12:00	280:09:03 / 280:21:03	03 / 280 CDT	-18.1				OCTOBER 6, 1988	STS 26	FINAL, REV C	9/16/88
CDT :280 FD : 03 MET :002											
STS	ZLV MNVR	MEAL						SLEEP			
D/O PREP REHEARSAL								PRE SLEEP			
PL	SSIP CRYSTALS/METALS	PCG PHOTO									
DAY/NIGHT ORBIT	33	34	35	36	37	38	39	40			
GSTDN COVERAGE	TDRS E W	ATTITUDE									
NOTES:											

GMT (D:H:M)	MET (D:H:M)	CDT (D:H:M)	FD/DOY	BETA	MOON	HOUSTON DATE	FLIGHT	EDITION	PUB. DATE
281:02:03 / 281:14:03	002:12:00 / 003:00:00	280:21:03 / 281:09:03	04 / 280 CDT	-16.4		OCTOBER 6, 1988	STS 26	FINAL, REV C	9/16/88
CDT :280	22	0 CDT:281	2						
FD : 03	12	15 FD: 04 16	17						
MET :002	13	14							MET:003
STS	SLEEP						FCS C/O		
	POST SLEEP								
PL									
DAY/NIGHT									
ORBIT									
GSTDN									
COVERAGE									
TDRS									
ATTITUDE									
NOTES:									

9/16/88

CAP/38/FIN C

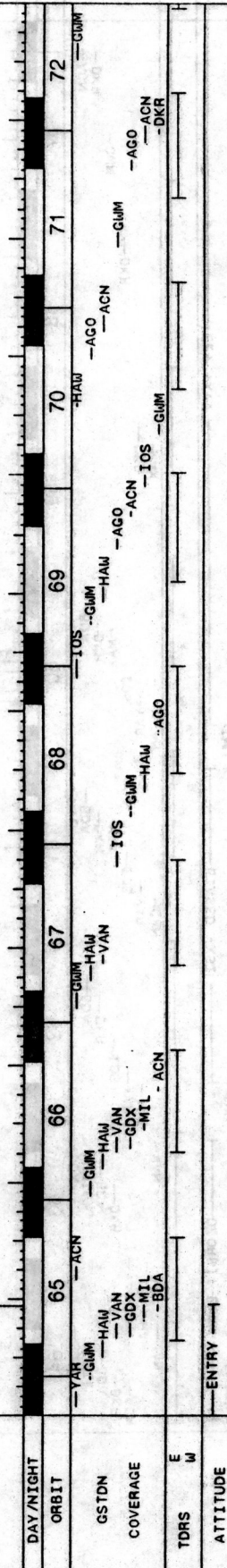
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281:14:03 / 282:02:03	003:00:00 / 003:12:00	281:09:03 / 281:21:03	04 / 281 CDT	-14.7		OCTOBER 7, 1988	STS 26	FINAL, REV C	9/16/88
CDT :281									
FD : 04									
MET :003									
STS	MEAL	CABIN STOW							
PL									
DAY/NIGHT									
ORBIT									
GSTON									
COVERAGE									
TDRS									
ATTITUDE									
NOTES:	● ZLV, +XVV ● ZLV, +XVV ● DTO 0623 CABIN AIR MONITORING ▶ ADSF 4 ACT								

[illegible]

GMT (D:H:M)	MET (D:H:M)	CDT (D:H:M)	FD/DOY	BETA	MOON	HOUSTON DATE	FLIGHT	EDITION	PUB. DATE
282:14:03/ 283:02:03	004:00:00/ 004:12:00	282:09:03/ 282:21:03	05/ 282 CDT	-11.1		OCTOBER 8, 1988	STS 26	FINAL, REV C	9/16/88
CDT :282									
FD : 05									
MET :004									
STS									
PL									
DAY/NIGHT									
ORBIT									
GSTDN									
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TDRS									
ATTITUDE									
NOTES:									

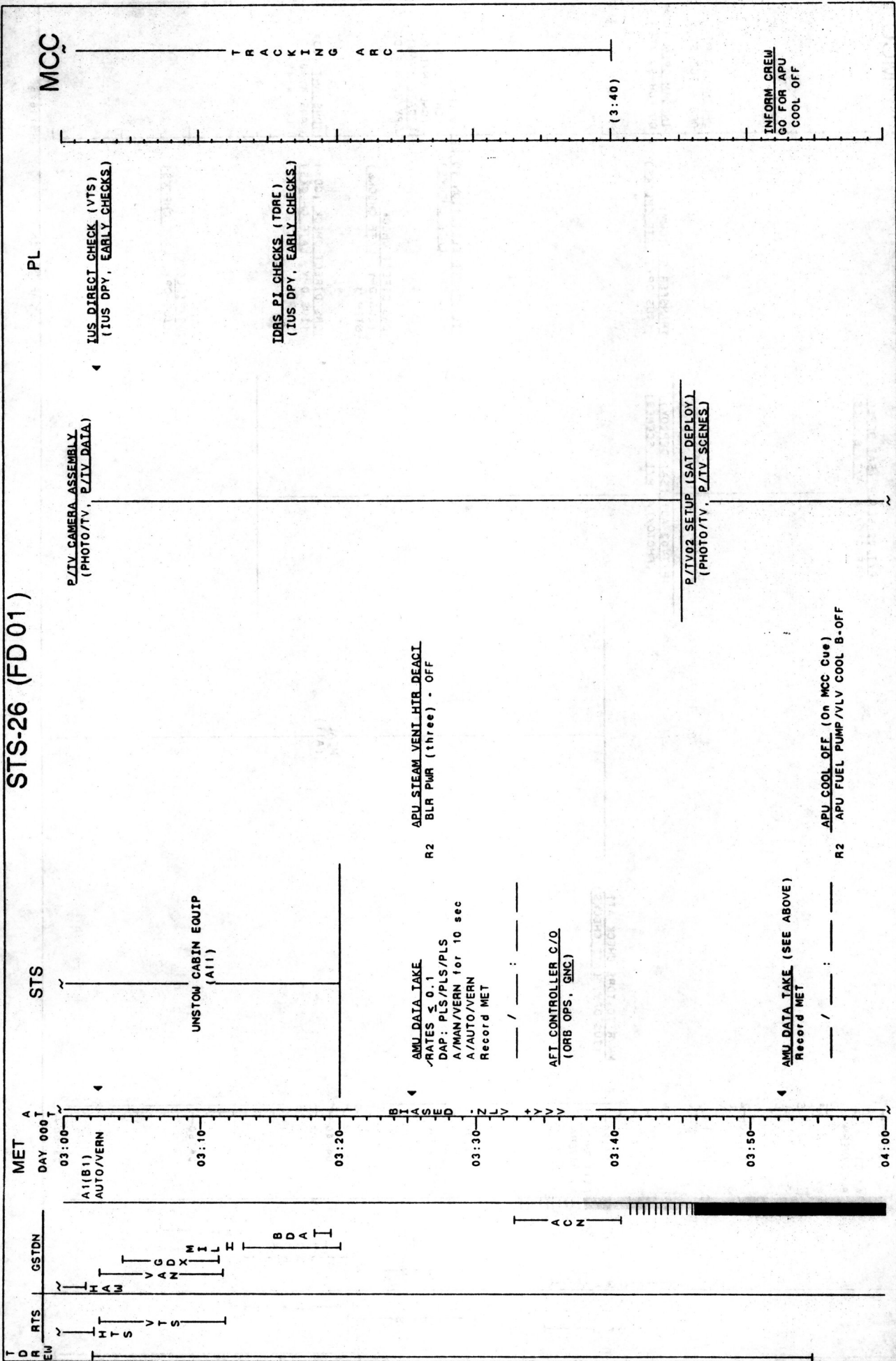
EDW LANDING
(4/00:57)



DETAILED TIMELINE

DETAILED
TIMELINE

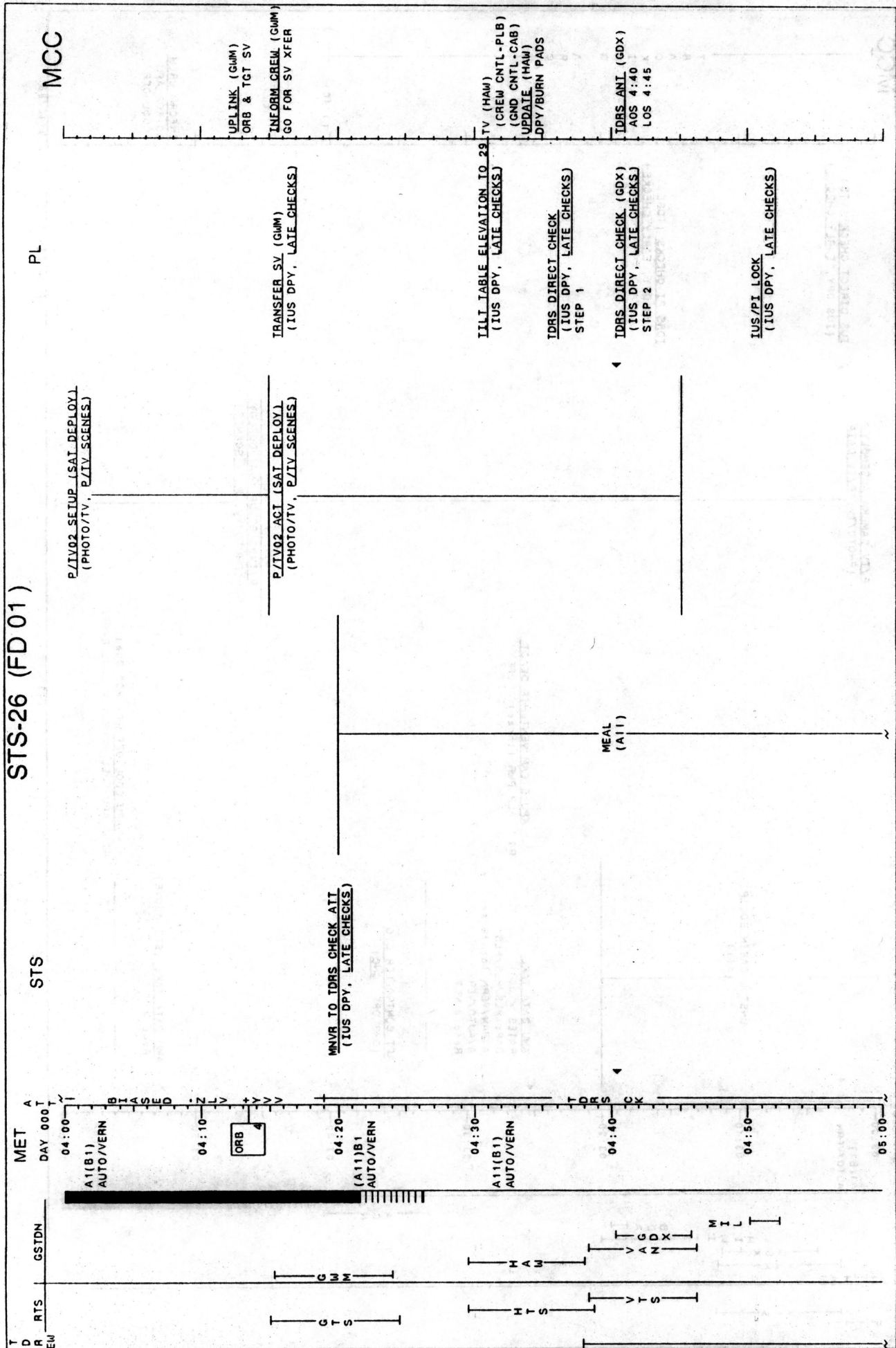
STS-26 (FD 01)



9/16/88

CAP/26/FIN C

STS-26 (FD 01)



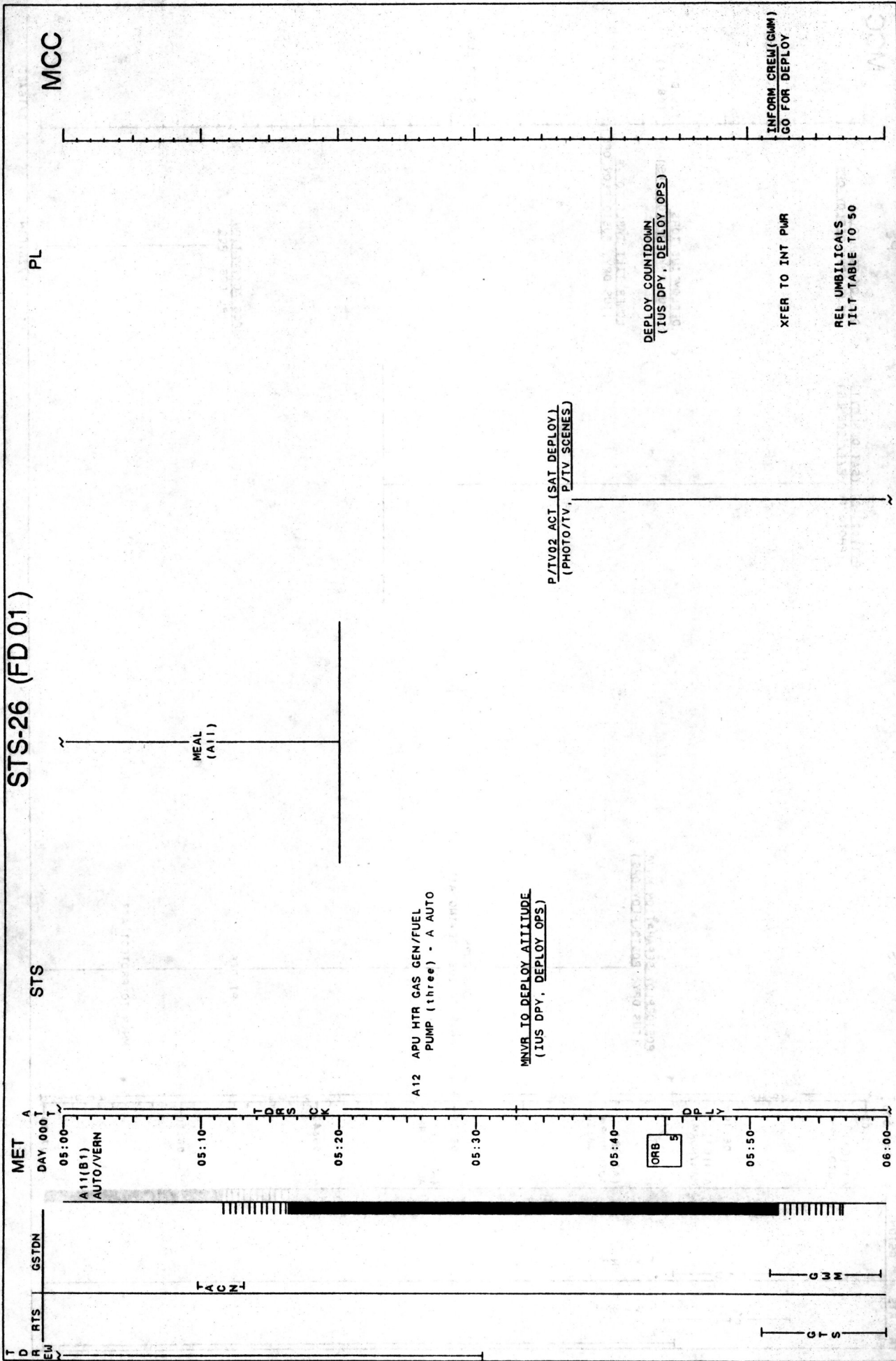
9/16/88

CAP/26/FIN C

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CL-26 (FD 01)

STS-26 (FD 01)



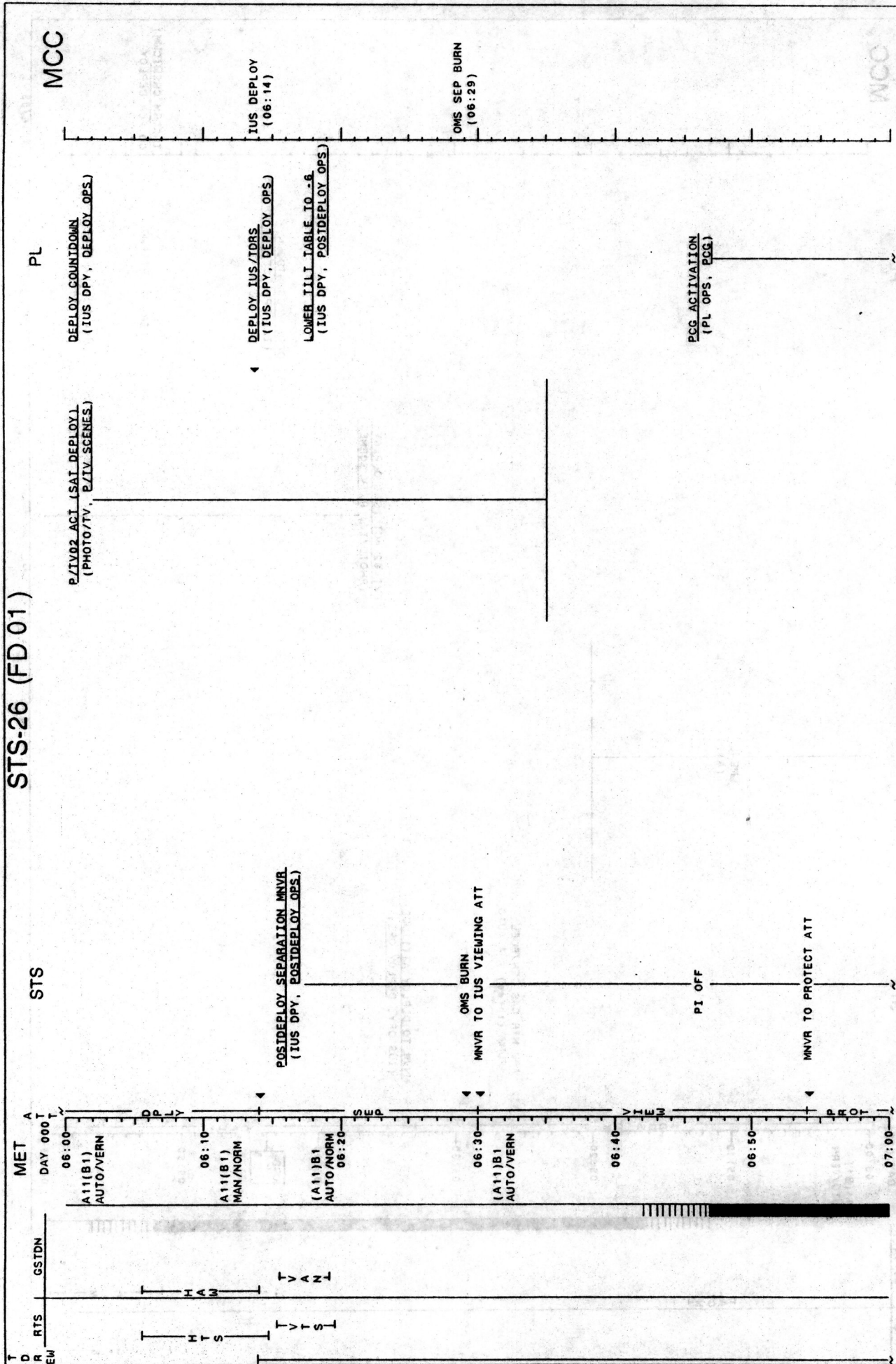
9/16/88

CAP/26/FIN C

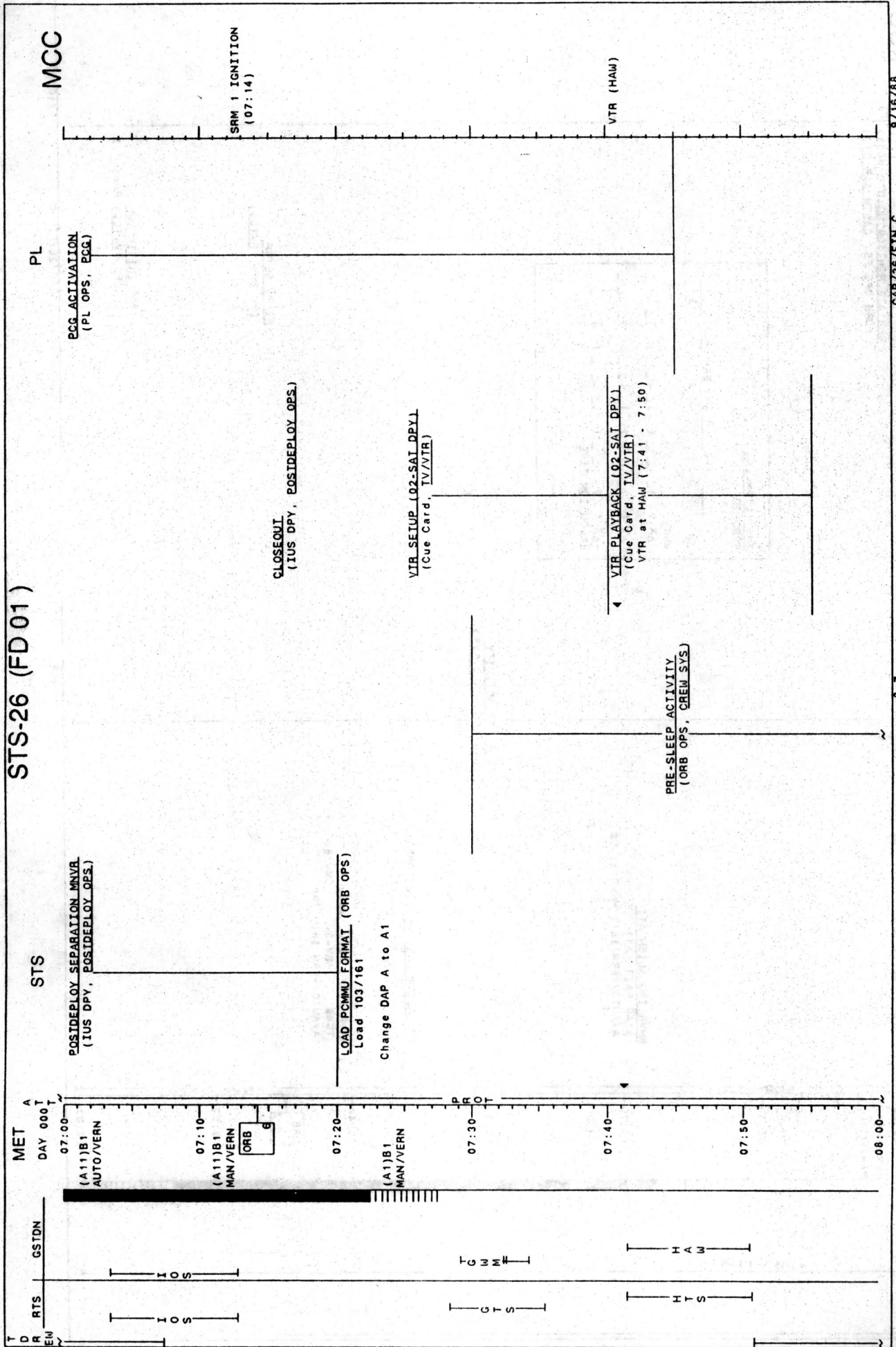
3-5

21236 (FD 01)

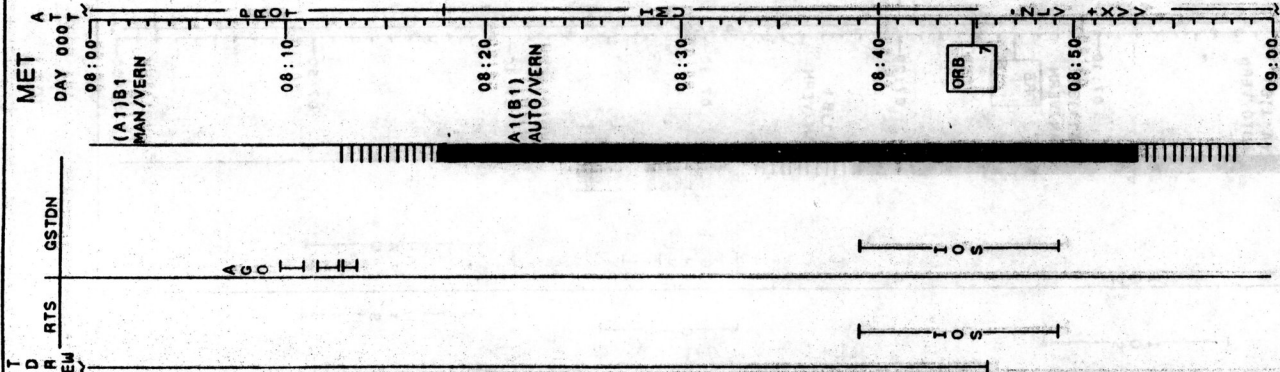
STS-26 (FD 01)



STS-26 (FD 01)



STS-26 (FD 01)



STS

PL

MCC

CAB AIR MONITORING(DTO 0623)
(ORB OPS FS, CABIN AIR DIO)

IMU ALIGN LOG
REQD ID: _____

-Y _____ -Z _____ ANG ERR _____

ANG: _____

ΔX () _____ () _____ () _____

ΔY () _____ () _____ () _____

ΔZ () _____ () _____ () _____

EXECUTION TIME: _____

--- / --- : --- : ---

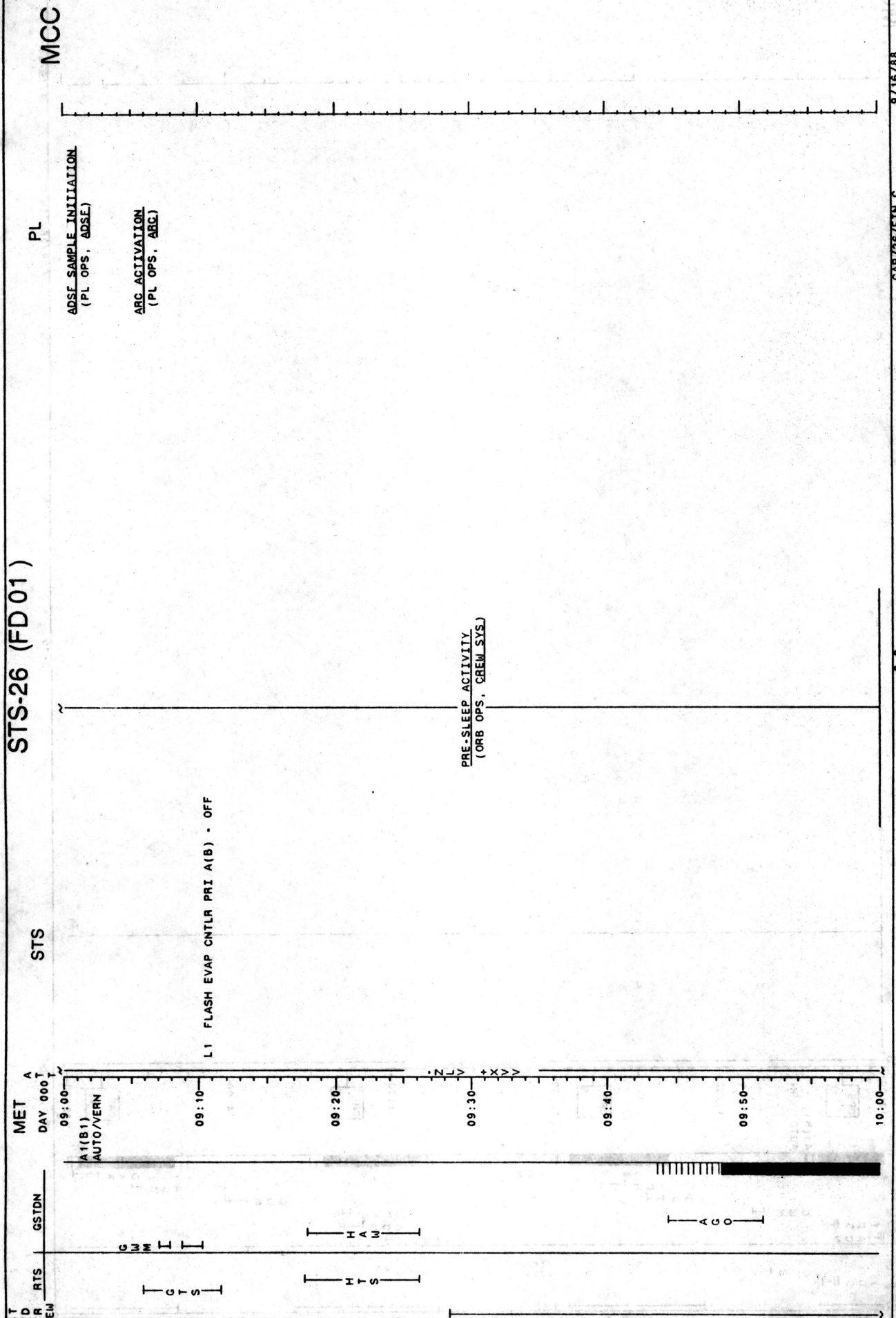
PRE-SLEEP ACTIVITY
(ORB OPS, CREW SYS)

IMU ALIGN - S TRK (ORB OPS)

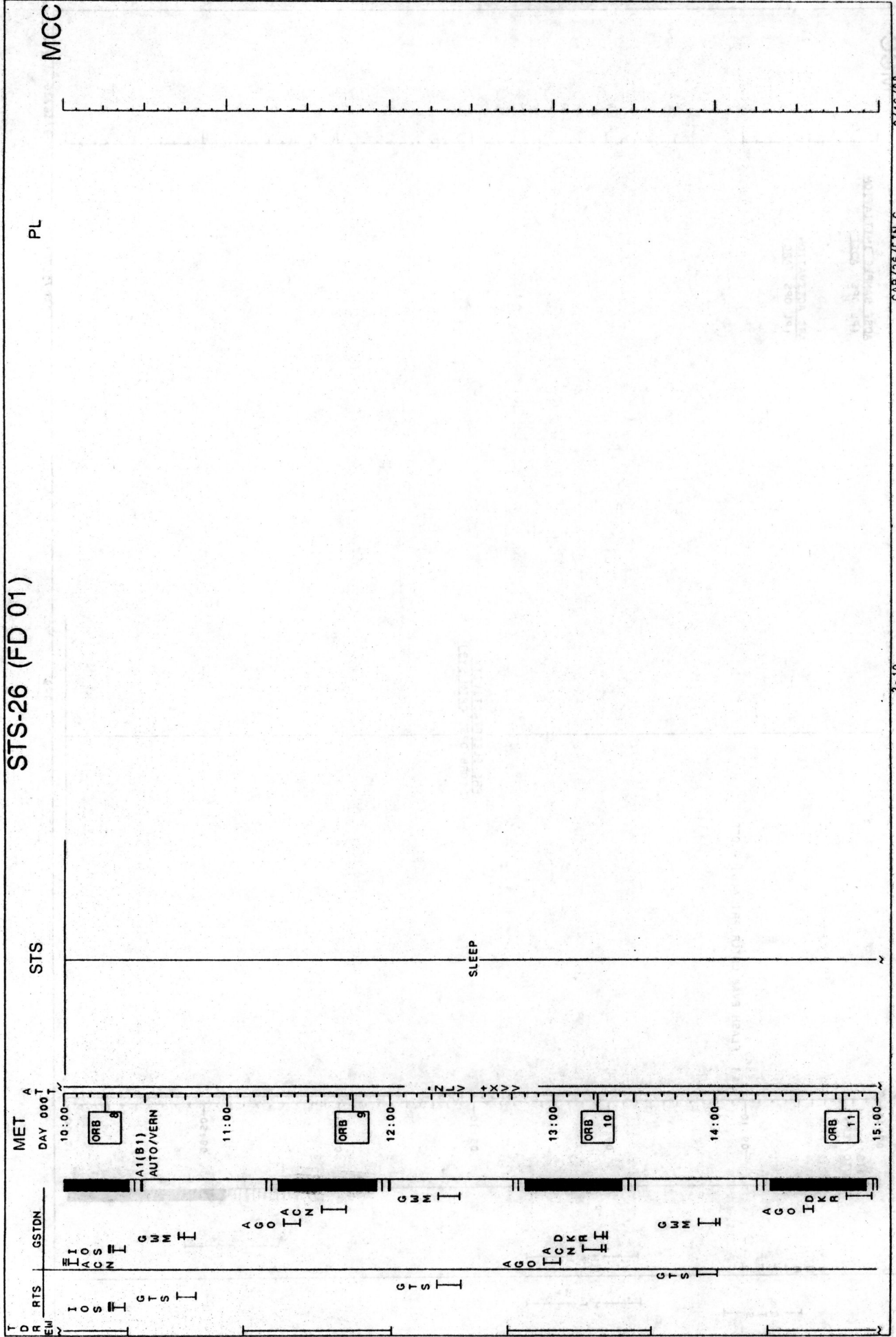
PVTOS SETUP
(PL OPS, PVTOS)

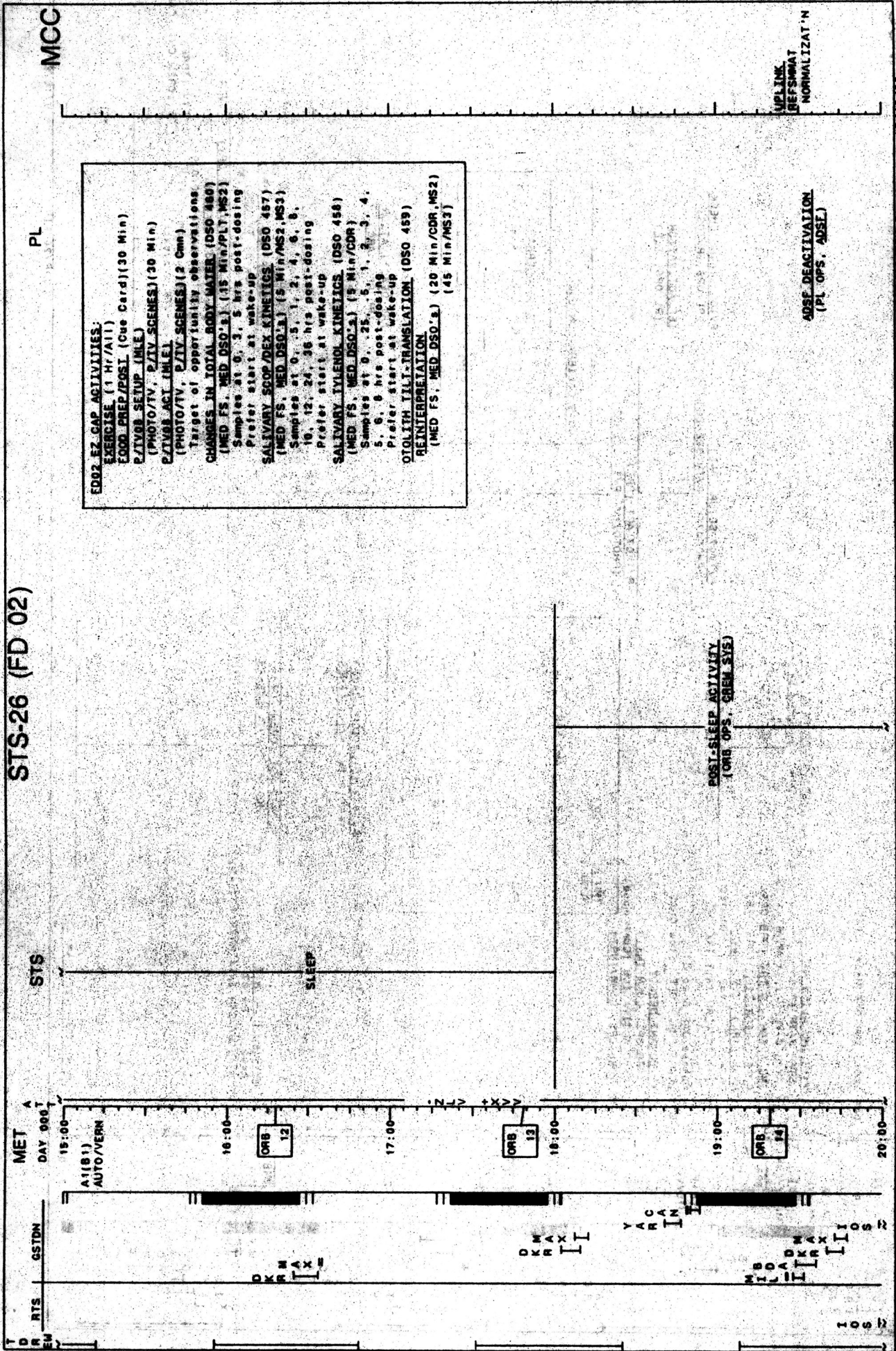
PVTOS INITIATION
(PL OPS, PVTOS) Runs 1.5

STS-26 (FD 01)



STS-26 (FD 01)





MCC

PL

UPLINK
REFS/MAT
NORMALIZATION

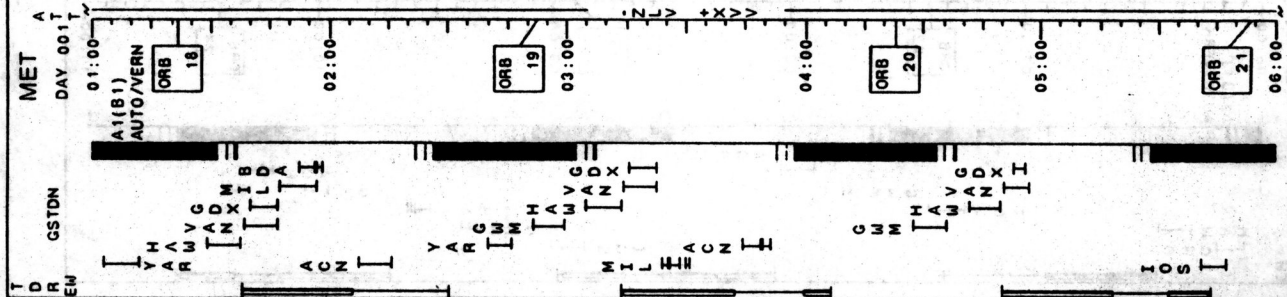
ADSE DEACTIVATION
(PL OPS, ADSE)

9/16/88

CAP/26/PLN C

3-11

STS-26 (FD 02)



MCC

P/TV07 SETUP (PPE)
(PHOTO/TV, P/TV SCENES)

P/TV07 ACT (PPE)
(PHOTO/TV, P/TV SCENES)

9/15/88

CAP/26/IN C

3-13

STS-26 (FD 02)

T D R EW
GSTDN
MET DAY 001

STS

PL

MCC

CAR AIR MONITORING (OTO 0823)
(ORB OPS FS, CABIN AIR DIO)

ECG PHOTO SESSION
(PL OPS, ECG)

MNVR IMU ALIGN AT1
STAR PAIR (A1)
A/AUTO/VERN Init MNVR at SS
IMU ALIGN - S TRK (ORB OPS)
MNVR (TRK) - ZLV - XXV
TG-2 BV-3 ON-180
A/AUTO/VERN Init TRK

PRE-SLEEP ACTIVITY
(ORB OPS, CREW SYS)

PVTOS INITIATION
(PL OPS, PVTOS) Runs 2.6
ADSE SAMPLE INITIATION
(PL OPS, ADSE)

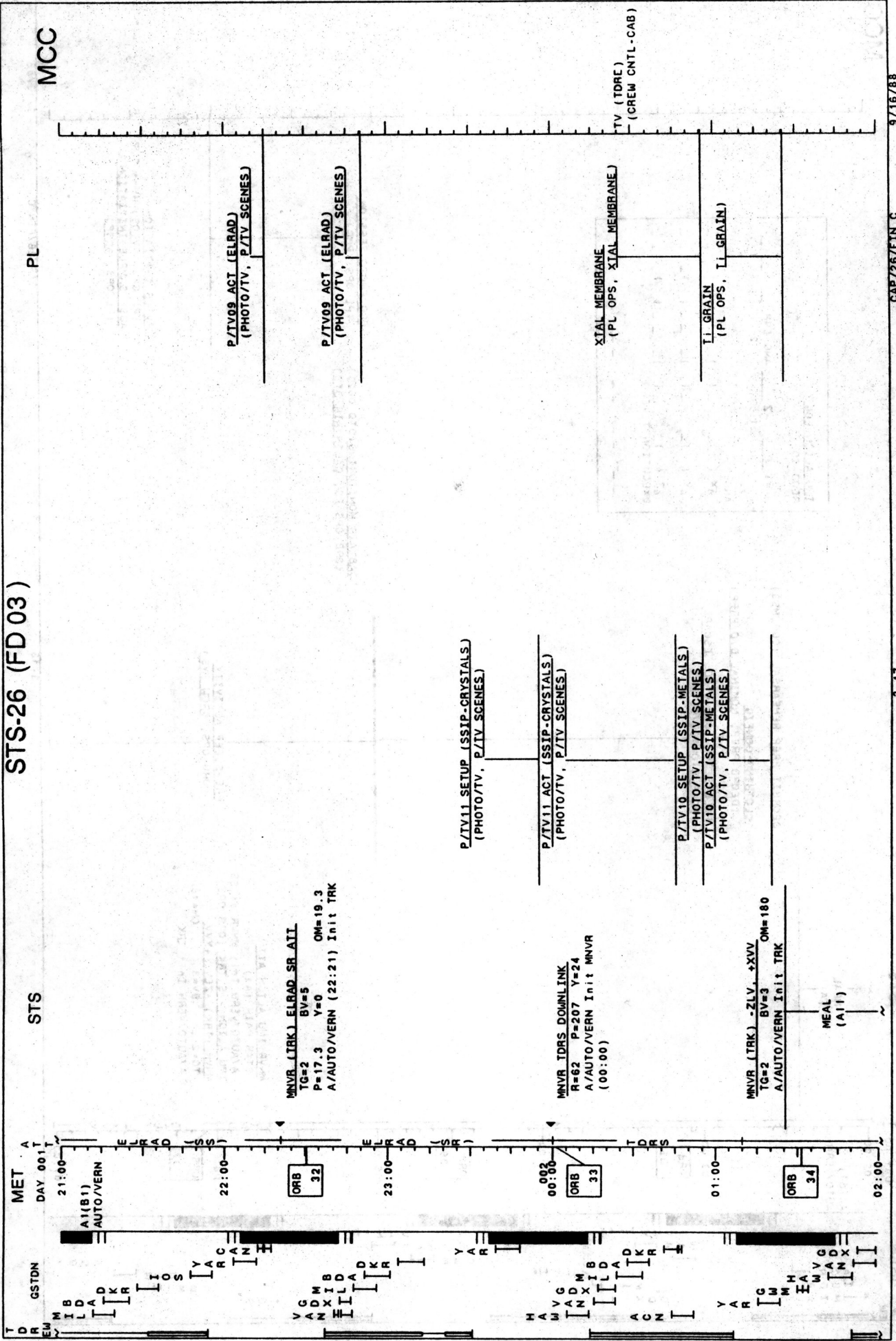
IMU ALIGN LOG		
REC'D ID:		
-Y	-Z	ANG ERR
1	2	3
ANG:		
ΔX ()	()	()
ΔY ()	()	()
ΔZ ()	()	()
EXECUTION TIME:		
-- / -- : -- : --		

SLEEP

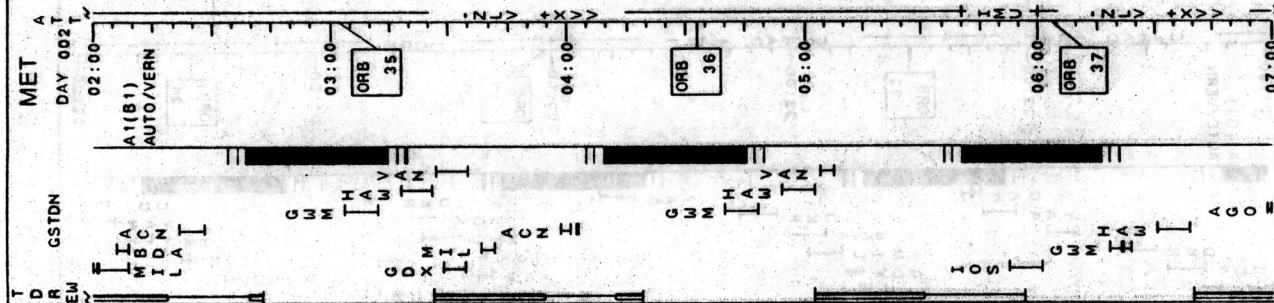
CON

97/16/88

STS-26 (FD 03)



STS-26 (FD03)



STS

PL

MCC

IMU ALIGN LOG
 RECD ID: -Y -Z - ANG ERR - - -
 ANG: ΔX () - - - () - - -
 ΔY () - - - () - - -
 ΔZ () - - - () - - -
 EXECUTION TIME: - - - - -
 - - - - -

DEORBIT PREP REHEARSAL (CDR, MS3)
 CLOTHING CONFIG
 (DEORB PREP, NOMINAL D/O PREP)
 POLE SETUP (CUE CARD, ESCAPE
 POLE STORAGE/SETUP)

CAB AIR MONITORING (DIO 0623)
 (ORB OPS FS, CABIN AIR DIO)

PCG PHOTO SESSION
 (PL OPS, PCG)

PRE-SLEEP ACTIVITY
 (ORB OPS, CREW SYS)

MNVR IMU ALIGN ATT
 STAR PAIR (A1)
 A/AUTO/VERN Init MNVR at SS
 IMU ALIGN - S TRK (ORB OPS)
 MNVR (TRK) - ZLV - XXV
 TG=2 BV=3 OM=180
 A/AUTO/VERN Init TRK

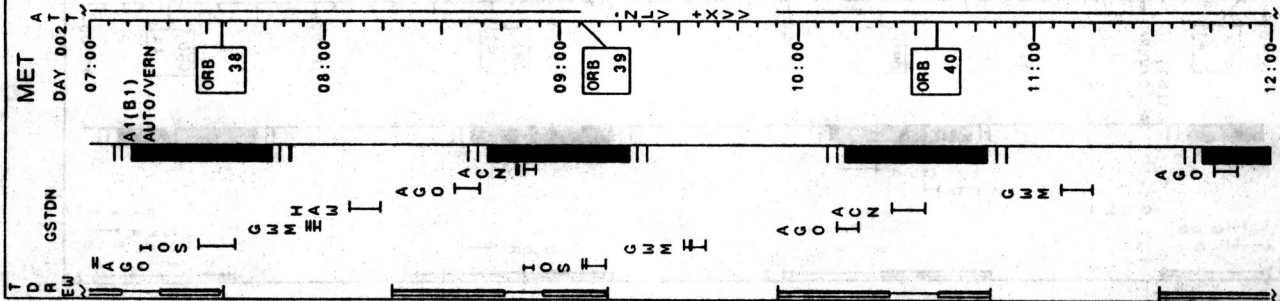
PVTOS INITIATION
 (PL OPS, PVTOS) Runs 3, 7
 ADSF SAMPLE INITIATION
 (PL OPS, ADSF)

3-18

9/16/88

CAP/26/FIN C

STS-26 (FD 03)

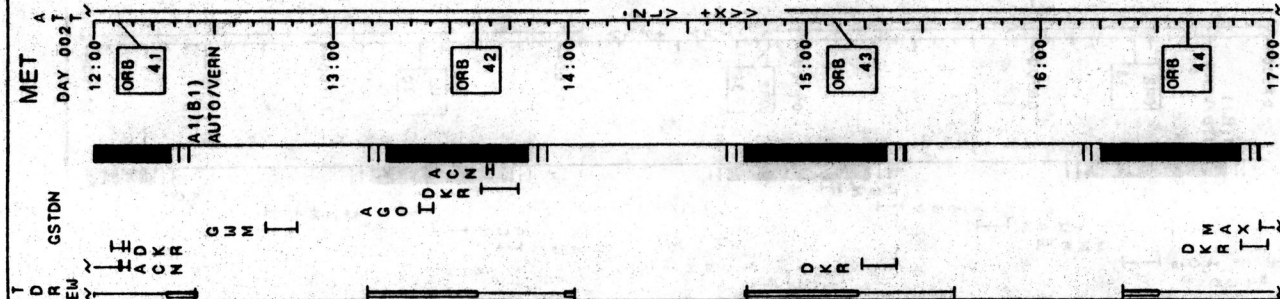


PL

STS

MCC

STS-26 (FD04)



STS

MET

DAY 002

GSTDN

T D R EN

PL

MCC

FD04 EZ CAP ACTIVITIES:

EXERCISE (1 Hr/Alt)

FOOD PREP/POST (Cue Card)(30 Min)

P/TV08 SETUP (MILE)

(PHOTO/TV, P/TV SCENES)(30 Min)

P/TV08 ACT (MILE)

(PHOTO/TV, P/TV SCENES)(2 Cmn)

Target of opportunity observations

SALIVARY TYLENOL KINETICS (DSO 458)

(MED FS, MED DSO's) (5 Min/CDR)

Samples at 0, .25, .5, 1, 2, 3, 4,

5, 6, 8 hrs post-dosing

Prefer start at wake-up

OTOLITH ILLI-TRANSLATION (DSO 459)

REINTERPRETATION

(MED FS, MED DSO's) (20 Min/CDR, MS2)

(45 Min/MS3)

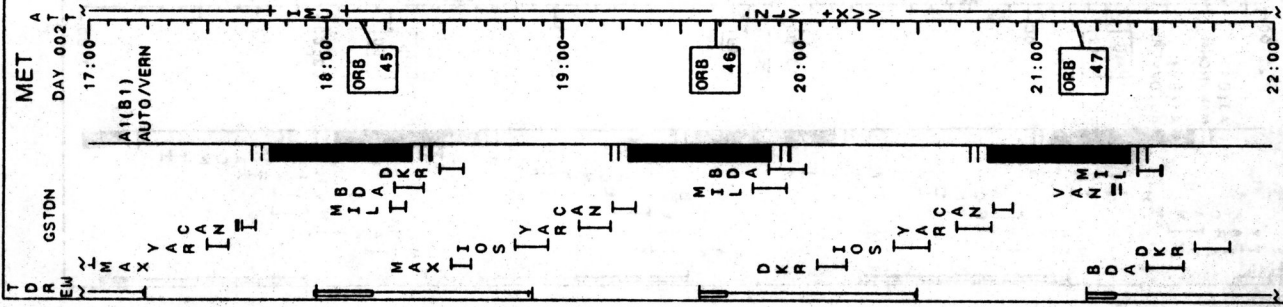
CHANGES IN TOTAL BODY WATER (DSO 460)

(MED FS, MED DSO's) (15 Min/PLT, MS2)

Samples at 0, 3, 5 hrs post-dosing

Prefer start at wake-up

STS-26 (FD04)



STS

PL

MCC

ADSE DEACTIVATION
(PL OPS, ADSE)

IMU ALIGN LOG
READ ID:

-Y	-Z	ANG ERR	---
1	2	3	

ANG:

ΔX	()	()	()
ΔY	()	()	()
ΔZ	()	()	()

EXECUTION TIME:

---/---:---:---

POST-SLEEP ACTIVITY
(ORB OPS, CREW SYS)

MNVR IMU ALIGN AIT
STAR PAIR (A1)
A/AUTO/VERN Init MNVR at SS
IMU ALIGN - S TRK (ORB OPS)
MNVR (TRK) - ZLV, +XV
TG=2 BV=3 OM=180
A/AUTO/VERN Init TRK

PVLOS STATUS CHECK
(PL OPS, PVLOS)

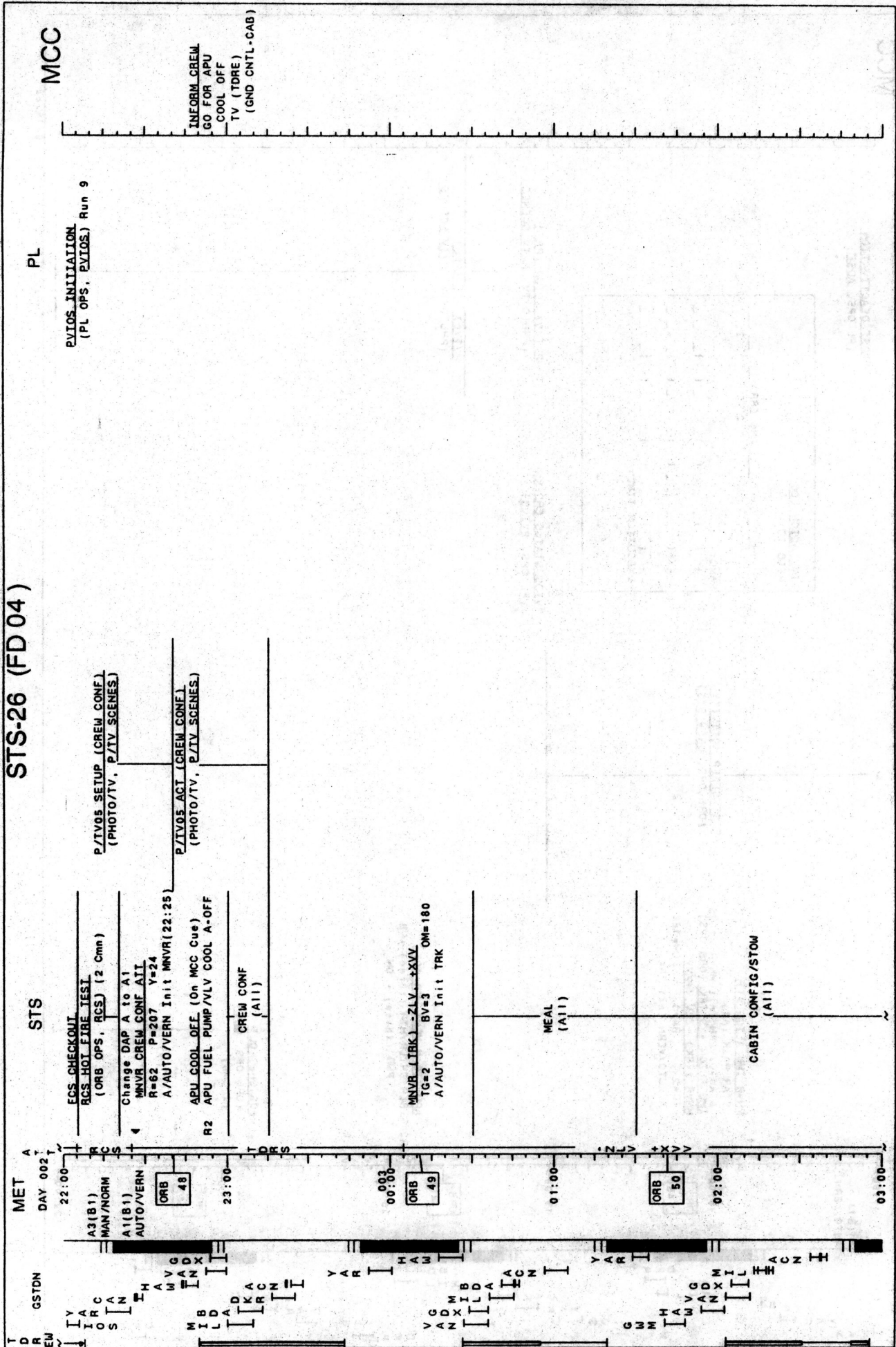
P/TV07 SETUP (PPEL)
(PHOTO/TV, P/TV SCENES)

P/TV07 ACT (PPE)
(PHOTO/TV, P/TV SCENES)

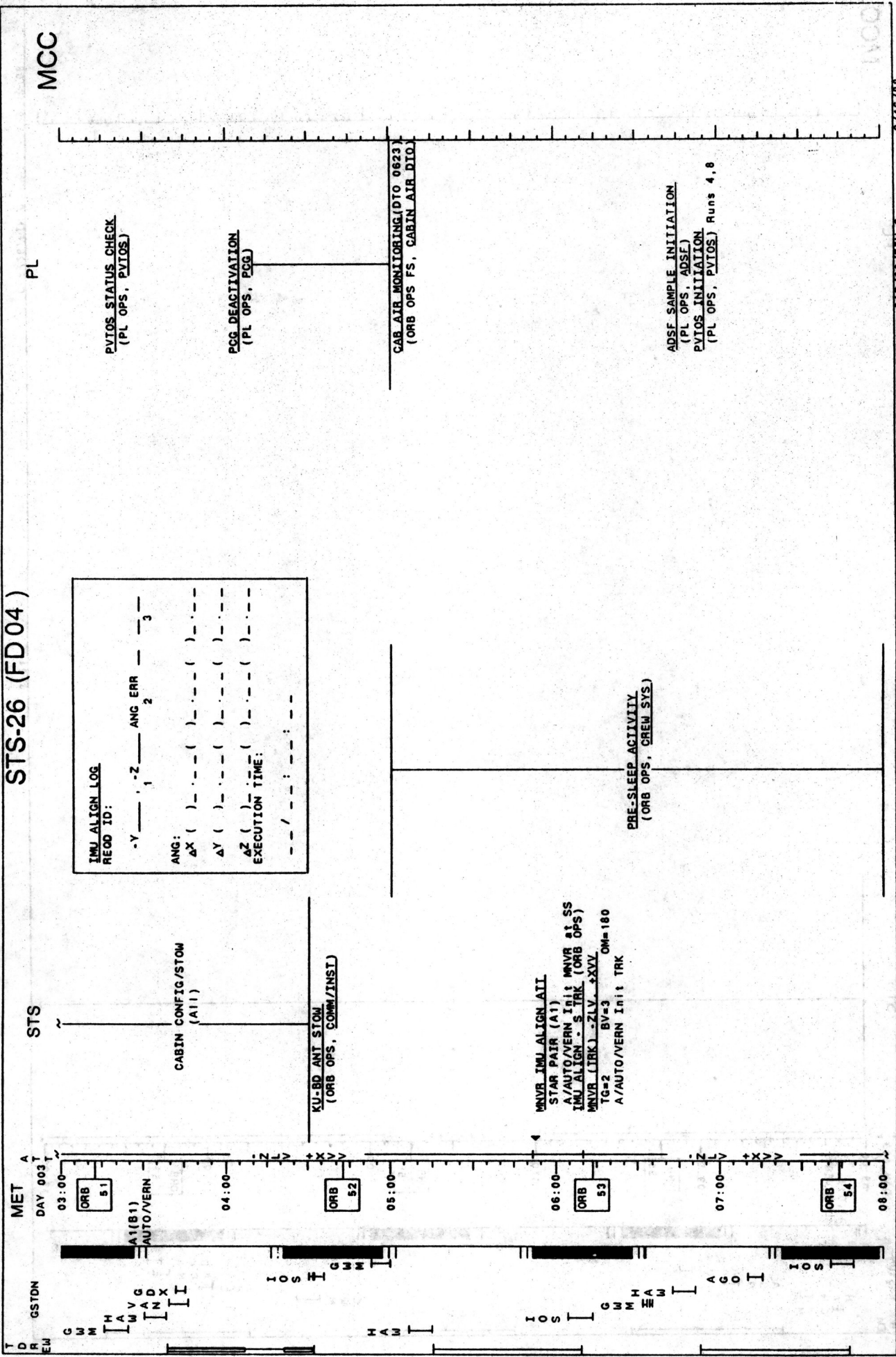
APU STEAM VENT HTR ACT
R2 BLR CNTLR/HTR (three) - B
PMR (three) - ON

FCS CHECKOUT
(ORB OPS, GNC)
(2 Cmm)

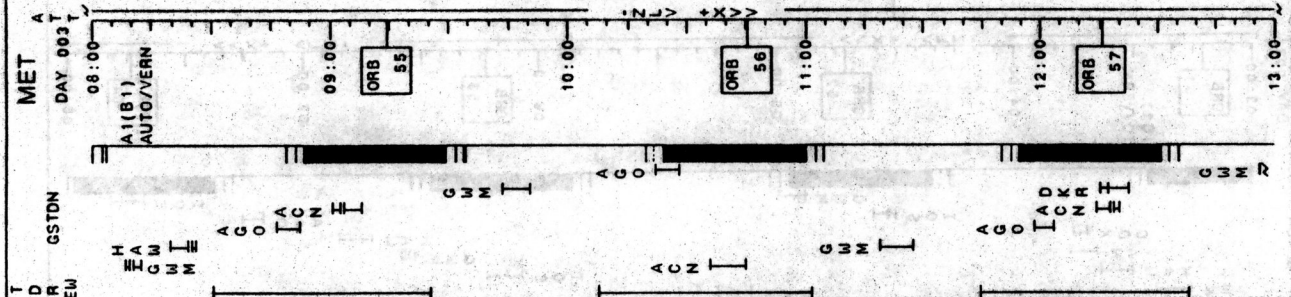
STS-26 (FD 04)



STS-26 (FD04)



STS-26 (FD 04)



STS

SLEEP

PL

MCC

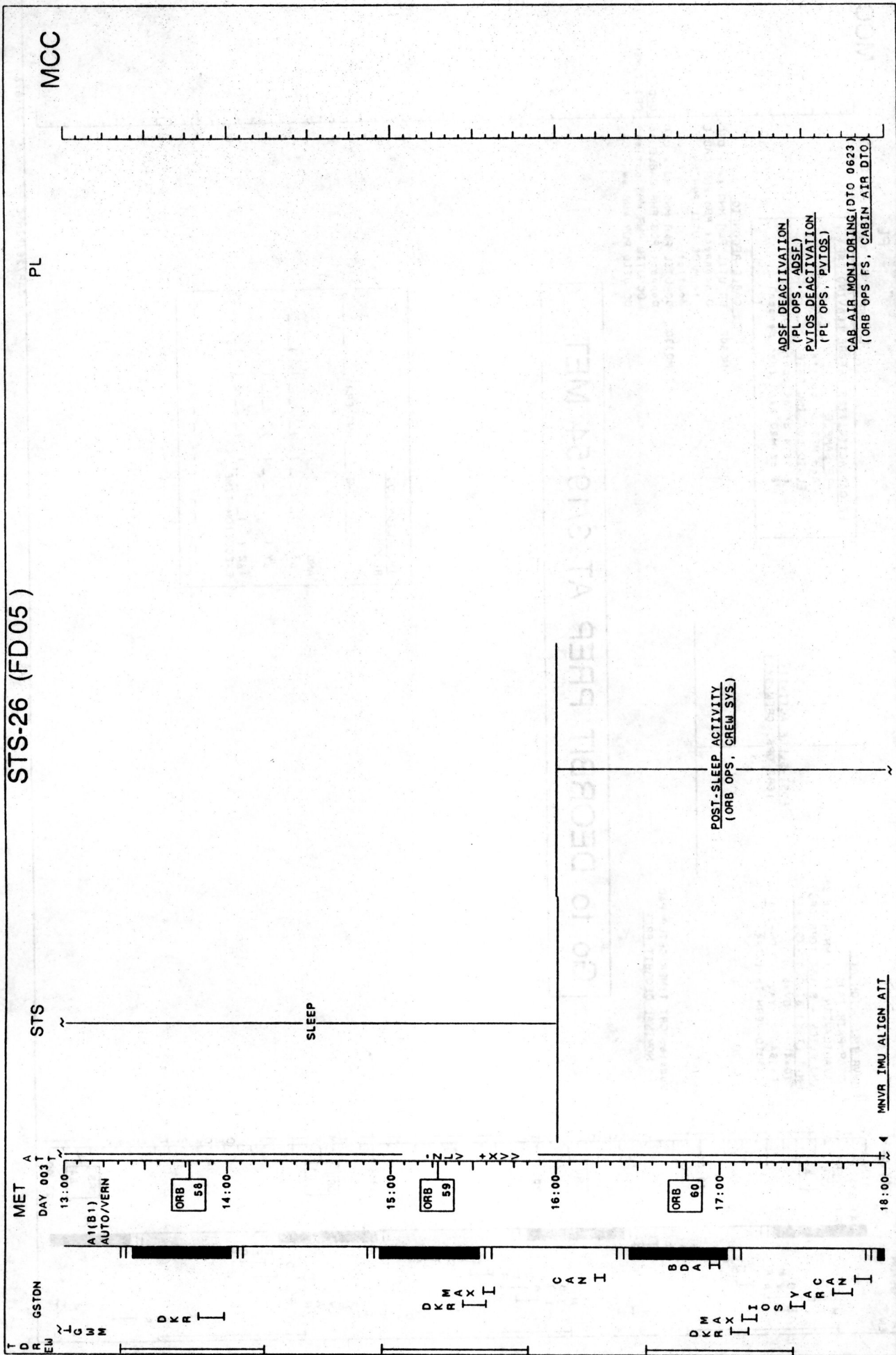
9/15/88

CAP/26/FIN C

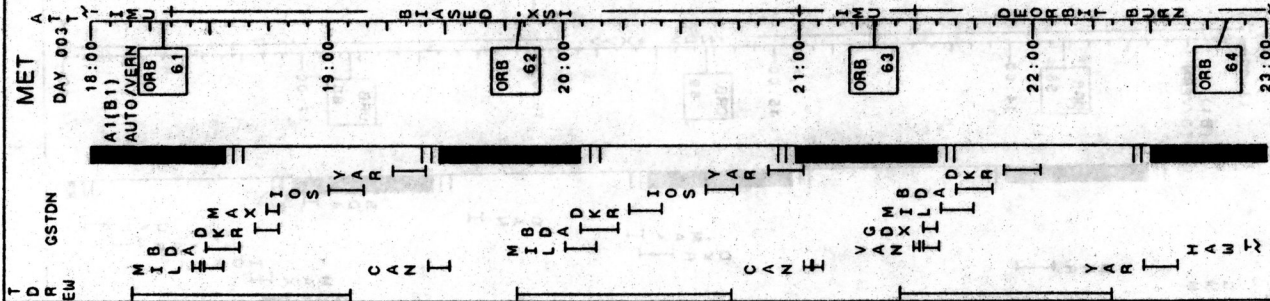
3-24

MCC

STS-26 (FD05)



STS-26 (FD05)



STS

PL

MCC

EZ CAP ACTIVITIES BEFORE LEAVING CAP:
 AIR SAMPLE
 Red MET on bottle
 FLUID LOADING PREP
 Fill 4 drink containers with
 8 oz H2O each (per person)

POST-SLEEP ACTIVITY
 (ORB OPS, CREW SYS)

PDB CABLE RECONNECT
 M030F DC UTIL PWR MNC sw - OFF
 Disconnect PDB PWR CABLE
 from DC UTIL PWR MNC
 outlet
 M0130 -DC UTIL PWR MNB sw - OFF
 Connect PDB PWR CABLE to
 DC UTIL PWR MNB outlet
 DC UTIL PWR MNB sw - ON

Update: CRT TIMER SETUP PAD
 in NOMINAL DEORBIT PREP

Go to DEORBIT PREP AT 3/19:54 MET

IMU ALIGN LOG

RECORD ID:

-Y -Z ANG ERR -3
 1 2
 ANG:
 ΔX () () () ()
 ΔY () () () ()
 ΔZ () () () ()
 EXECUTION TIME:
 - - - - -

CONTINGENCY TIMELINES

7A TDRS INSERTION.....	4A-1
8D TDRS INSERTION.....	4B-1
16A TDRS INSERTION.....	4C-1
18A TDRS (EVA) INSERTION.....	4D-1

CONTINGENCY
TIMELINES

1A TOPS INSERTION.....
1B TOPS INSERTION.....
1C TOPS INSERTION.....
1D TOPS INSERTION.....

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CONTINGENCY
TIMELINES

7A TDRS INSERTION

The 7A Insertion Timeline is designed to accommodate delaying deployment of TDRS to the 7A insertion opportunity. The nominal timeline is used until the GO/NO-GO point in the deploy countdown for the nominal (6A insertion) timeline. The timeline is based on the nominal Cycle 3R OMS-2 insertion.

[illegible]

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8D TDRS INSERTION

The 8D Insertion Timeline is designed to accommodate delaying deployment of TDRS for the 8D insertion opportunity. The 7A insertion timeline is used up to the GO/NO-GO decision point in the deploy countdown. An 8D insertion will be considered if a high probability of deploying exists or if a 16A insertion cannot be accommodated due to orbiter or IUS/TDRS problems. Additional assumptions are shown on page 4B-2. The timeline is based on the nominal Cycle 3R OMS-2 insertion.

8D TDRS

GMT (D:H:M)	MET (D:H:M)	CDT (D:H:M)	FD/DOY	BETA	MOON	HOUSTON DATE	FLIGHT	EDITION	PUB. DATE
278:14:03 / 279:02:03	000:00:00 / 000:12:00	278:09:03 / 278:21:03	01 / 278 CDT	-24.0	0	OCTOBER 4, 1988	STS 26	FINAL REV C	9/18/88
CDT :278	10	11	12	13	14	15	16	17	18
FD :001	9	2	3	4	5	6	7	8	9
MET :000	10	11	12	13	14	15	16	17	18
STS	THIS 8D TIMELINE MAKES THE FOLLOWING ASSUMPTIONS: -THE 7A B/U INSERTION T/L IS USED UNTIL 07:30 MET -THE IUS/IDRS ACTIVITIES ARE SHOWN TO GIVE THE LAST ACCEPTABLE TIME TO MAKE A SUCCESSFUL DEPLOY -THE ORBITER IS IN THE DEPLOY ATTITUDE FOR 7A INSERTION -THE TILT TABLE IS ELEVATED TO 29 DEGREES -THE MDL UPLINK SHOULD OCCUR OVER IOS, THIS ALLOWS HTS TO BE A BACKUP -A CLEAR TRACKING ARC IS AVAILABLE AS SHOWN IF A STATE VECTOR UPDATE IS REQUIRED -REM HEATERS ARE POWERED OFF IN THE 7A INSERTION T/L, THEY ARE NOT REQUIRED TO BE REPOWERED AGAIN FOR THIS 8D INSERTION -THE P/TV SETUP ACTIVITY WAS PERFORMED IN THE NOMINAL 6A T/L -THIS TIMELINE CANNOT WITHSTAND ANY LAUNCH SLIPS -THE MIDDECK PAYLOAD ACTIVITIES WILL BE REPLANNED								
PL	DEPLOY NO GO P/TV02 ACT								
DAY/NIGHT	ORBIT GSTDN COVERAGE RTS COVERAGE TDRS ATTITUDE								
NOTES:	8D INSERTION TRACKING ARC								

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16A TDRS INSERTION

The 16A Insertion Timeline is designed to accommodate delaying deployment of TDRS for the 16A insertion opportunity. The timeline assumes an attempt at 7A insertion failed at the GO/NO-GO decision point in the deploy countdown procedure. The timeline begins during crew post sleep and assumes the tilt table is stowed at 0°. Additional assumptions are shown on page 4C-2. The timeline is based on the nominal Cycle 3R OMS-2 insertion.

16A TDRS

[illegible]

GMT (D:H:M)	MET (D:H:M)	CDT (D:H:M)	FD/DOY	BETA	MOON	HOUSTON DATE	FLIGHT	EDITION	PUB. DATE
279:02:03 / 279:14:03	000:12:00 / 001:00:00	278:21:03 / 279:09:03	02 / 276 CDT	-22.7		OCTOBER 4, 1988	STS-26	FINAL, REV C	8/19/88
CDT :278		0 CDT:279	2						
FD :001		15 CDT:279	17						
MET :000		12	14						
STS	SLEEP								
PL									
DAY/NIGHT									
ORBIT									
GSTDN COVERAGE									
RTS COVERAGE									
TDRS E W									
ATTITUDE									
NOTES:	16A INSERTION TRACKING ARC TDRS CK ATT (MIL, MLX, BDA) TDRS DIR (MIL, MLX, BDA) PRE-DPY CK (TDRS)								

8/19/88

CAP/26/FIN C

4C-3

BACK TO THE NORMAL TIME LINE

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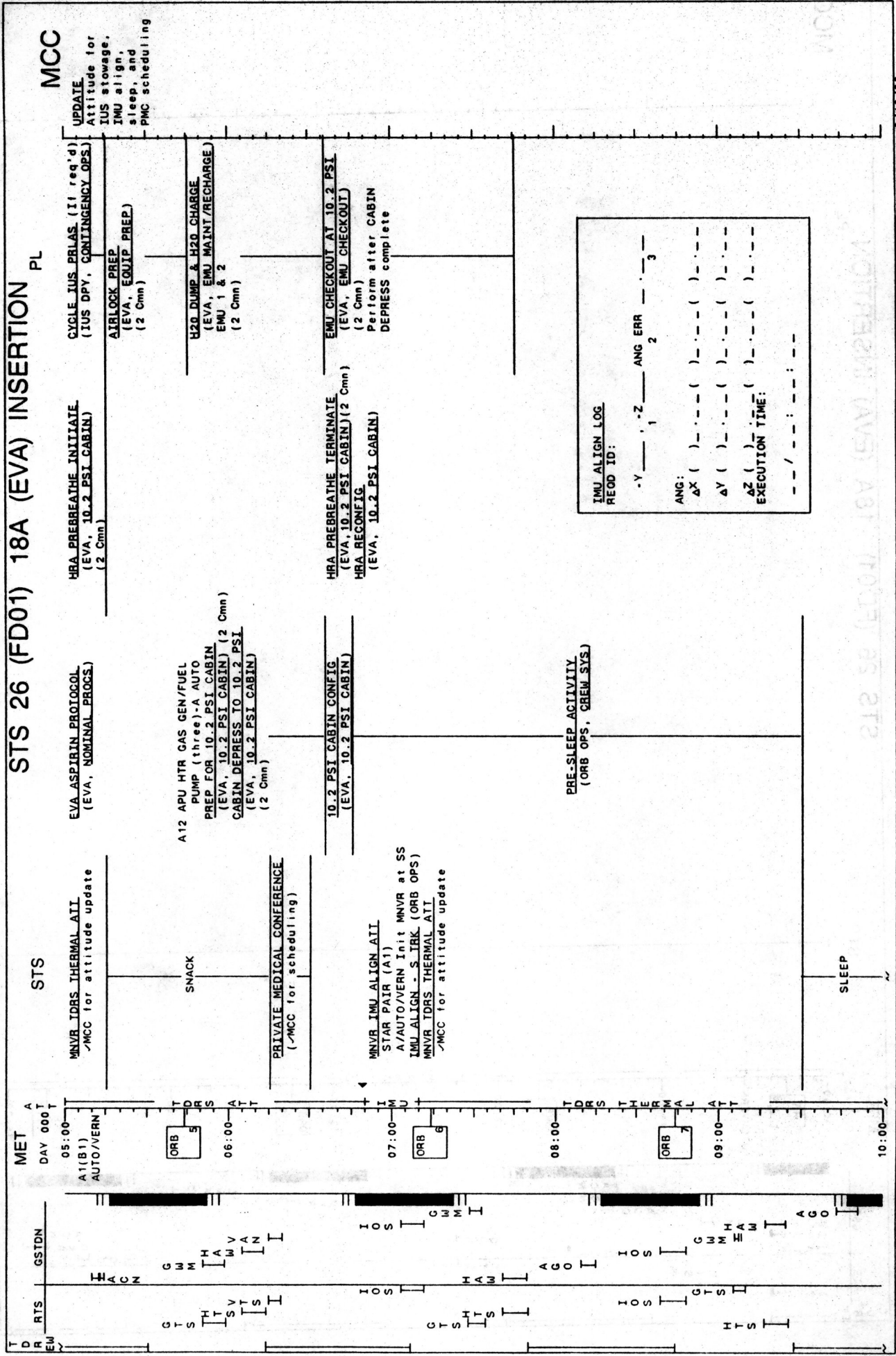
18A TDRS (EVA) INSERTION

The 18A Insertion Timeline is designed to accommodate delaying deployment of TDRS for the 18A insertion opportunity following an EVA to manually raise the tilt table. The decision to perform an EVA must be made by 0/05:00 MET in order to meet EVA preparation and equipment checkout constraints. The 10.2 psi cabin EVA protocol is used. There will be no Flight Day 1 EVA. The Contingency EVA Operations STS-26 Flight Supplement, Contingency EVA for Payload Failures Section, is the source document for the 18A insertion timeline. Additional assumptions are shown on pages 4D-2 and 4D-3. The timeline is based on the nominal Cycle 3R OMS-2 insertion.

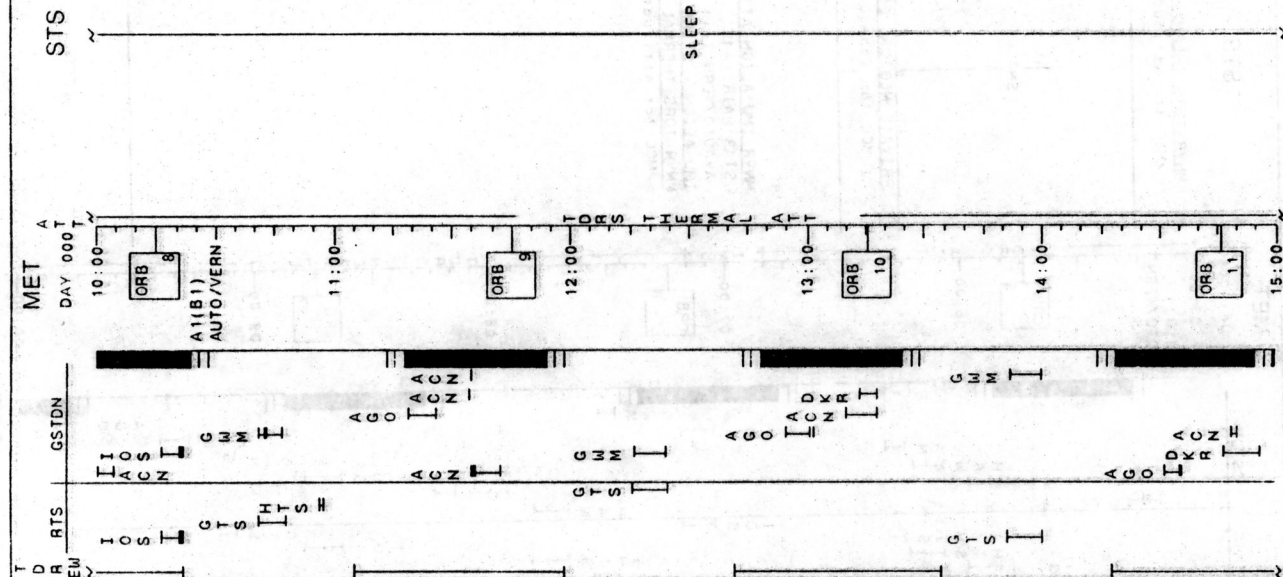
18A TDRS

GMT (D:H:M)	MET (D:H:M)	CDT (D:H:M)	FD/DOY	BETA	MOON	HOUSTON DATE	FLIGHT	EDITION	PUB. DATE
278:14:03 / 279:02:03	000:00:00 / 000:12:00	278:09:03 / 278:21:03	01 / 278 CDT	-24.0	1	OCTOBER 4, 1988	STS-28	FINAL, REV C	9/16/88
CDT :278									
FD :001									
MET :000									
STS	18A TIMELINE ASSUMPTIONS: "GO FOR EVA" FOLLOWS TILT TABLE FAILURE DURING RAISING TO 29 DEG "LATEST DEPRESS START" IS THE LATEST TIME TO START A CABIN DEPRESS AND STILL MEET THE FD2 SCHEDULE. "LATEST SLEEP START" WILL ALLOW AN 18A INSERTION WITH MINIMUM 6 HOURS CREW SLEEP. - PL SYS, PL MAL 1.1c PROCEDURE FOLLOWING TILT TABLE FAILURE IS UNSUCCESSFUL. - FLIGHT RULE A2-45 IS SATISFIED. - REFER TO FLIGHT RULE 13-11 FOR PREBREATHE PROTOCOL								
PL	GO FOR EVA 18A TIMELINE ASSUMPTIONS: "GO FOR EVA" FOLLOWS TILT TABLE FAILURE DURING RAISING TO 29 DEG "LATEST DEPRESS START" IS THE LATEST TIME TO START A CABIN DEPRESS AND STILL MEET THE FD2 SCHEDULE. "LATEST SLEEP START" WILL ALLOW AN 18A INSERTION WITH MINIMUM 6 HOURS CREW SLEEP. - PL SYS, PL MAL 1.1c PROCEDURE FOLLOWING TILT TABLE FAILURE IS UNSUCCESSFUL. - FLIGHT RULE A2-45 IS SATISFIED. - REFER TO FLIGHT RULE 13-11 FOR PREBREATHE PROTOCOL								
DAY/NIGHT	DAY/NIGHT ORBIT GSTDN COVERAGE RTS COVERAGE TDRS ATTITUDE								
NOTES:	18A (EVA) INSERTION ● IF NOT PREVIOUSLY OPERATED								

[illegible]

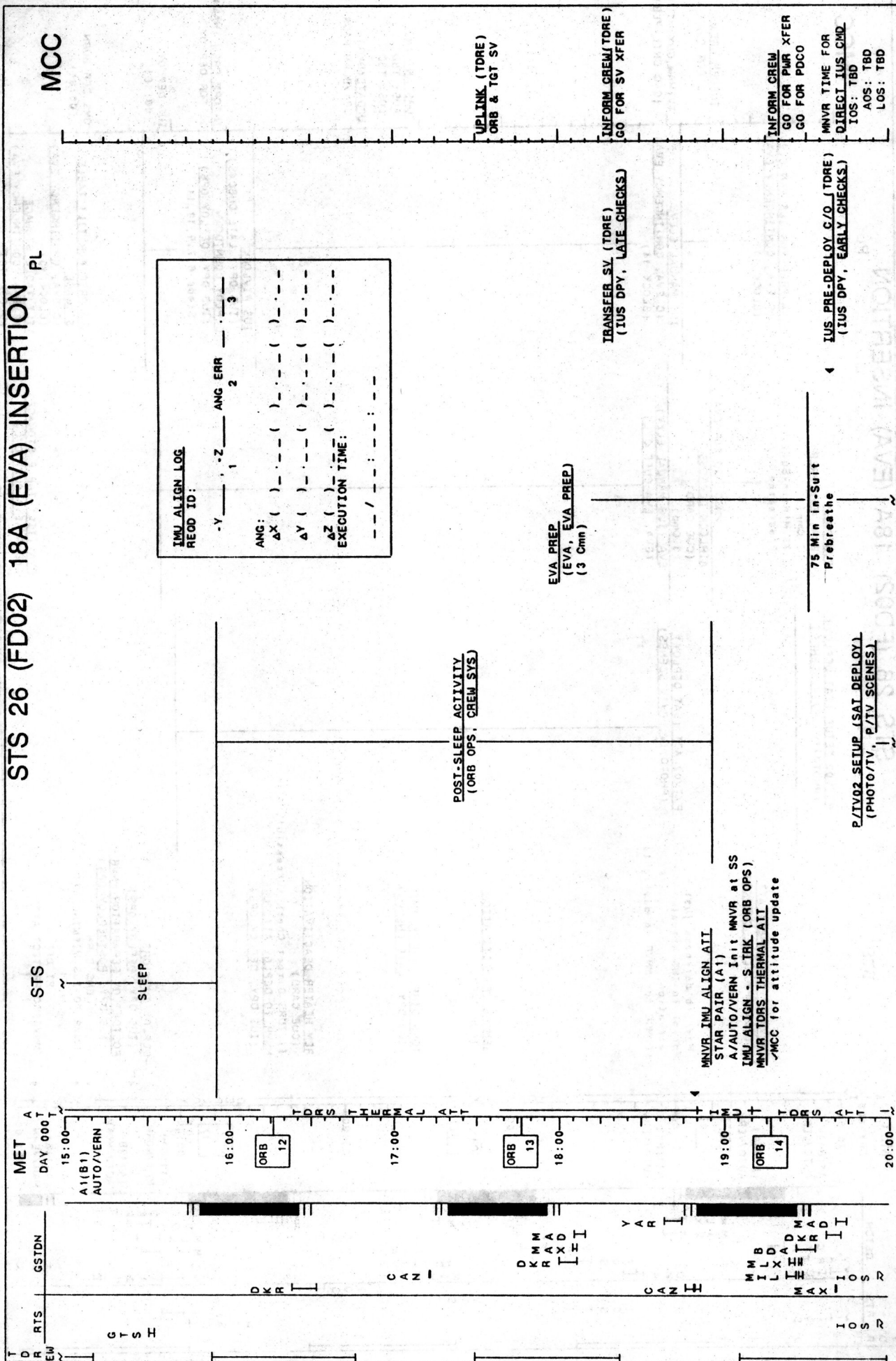


STS 26 (FD01) 18A (EVA) INSERTION PL

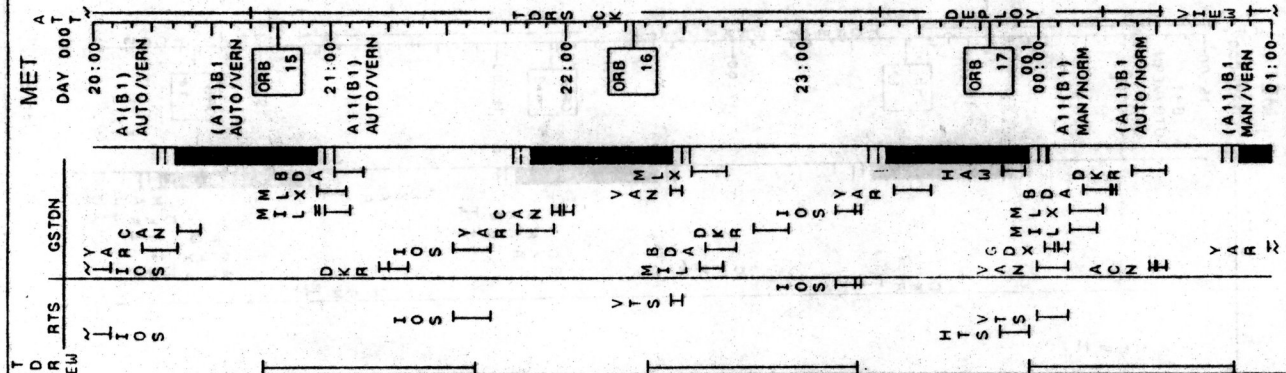


MCC

STS 26 (FD02) 18A (EVA) INSERTION PL



STS 26 (FD02) 18A (EVA) INSERTION PL



STS

P/TV02 SETUP (SAT DEPLOY) (PHOTO/TV, P/TV SCENES)

EVA PREP (EVA, EVA PREP) (3 Cmn) 75 Min in-Suit Prebreathe

EVA PREP FOR IUS MAN ELEV (C EVA, CONTINGENCY EVA) (BLOCK 11)

~MCC: GO FOR EVA

AIRLOCK DEPRESS (CUE CARD)

EVA (IUS MANUAL ELEV) (EVA, EVA CUFF C/L)

P/TV02 ACT (SAT DEPLOY) (PHOTO/TV, P/TV SCENES)

IUS MANUAL ELEV (C EVA, CONTINGENCY EVA) (BLOCK 14)

INFORM CREW GO FOR EVA

TV (HAW, GDX, MIL) (GND CNTL-PLB)

TDRS ANI (MIL) AOS: TBD LOS: TBD UPDATE (DKR) DPV/BURN PADS

IUS PI/LOCK (IUS DPV, LATE CHECKS) DEPLOY COUNTDOWN (IUS DPV, DEPLOY OPS) Steps 6, 7, 9, 10, 11

INFORM CREW (HAW) GO FOR DEPLOY

IUS DEPLOY (00:16)

OMS SEP BURN (00:31)

EVA PREP FOR IITL TABLE STORAGE (C EVA, CONTINGENCY EVA) (BLOCK 13)

IITL TABLE STORAGE (C EVA, CONTINGENCY EVA) (BLOCK 16)

EVA (IITL TABLE STORAGE) (EVA, EVA CUFF C/L)

9/16/88

40-8

CAP/26/FIN C

STS 26 (FD02) 18A (EVA) INSERTION

PL

MCC

STS 26 (FD02) 18A (EVA) INSERTION

PL

EVA (UTIL TABLE STOMAGEL (EVA, EVA CUFF C/L), AIRLOCK REPRESS (CUE CARD) (3 Cmn), POST EVA (EVA, POST EVA) (3 Cmn)

VTR PLAYBACK (02-SAT DPV) (Cue Card, TV/VTR) VTR at MIL

VTR (MIL)

SRM 1 IGNITION (01:17)

01:00 MET

01:17 MET

01:25 MET

02:00 MET

03:00 MET

04:00 MET

05:00 MET

06:00 MET

ORB 18

ORB 19

ORB 20

ORB 21

Change DAP A to A1

MNVR TDORS ATT

R=62 P=207 Y=24

A/AUTO/VERN (01:25) Init MNVR

KU-BD ANT DEPLOY

(ORB OPS, COMM/INST)

KU-BD ACTIVATION (COMM mode)

(ORB OPS, COMM/INST)

MEAL (A11)

MNVR (TRK) -ZLV, +XVV

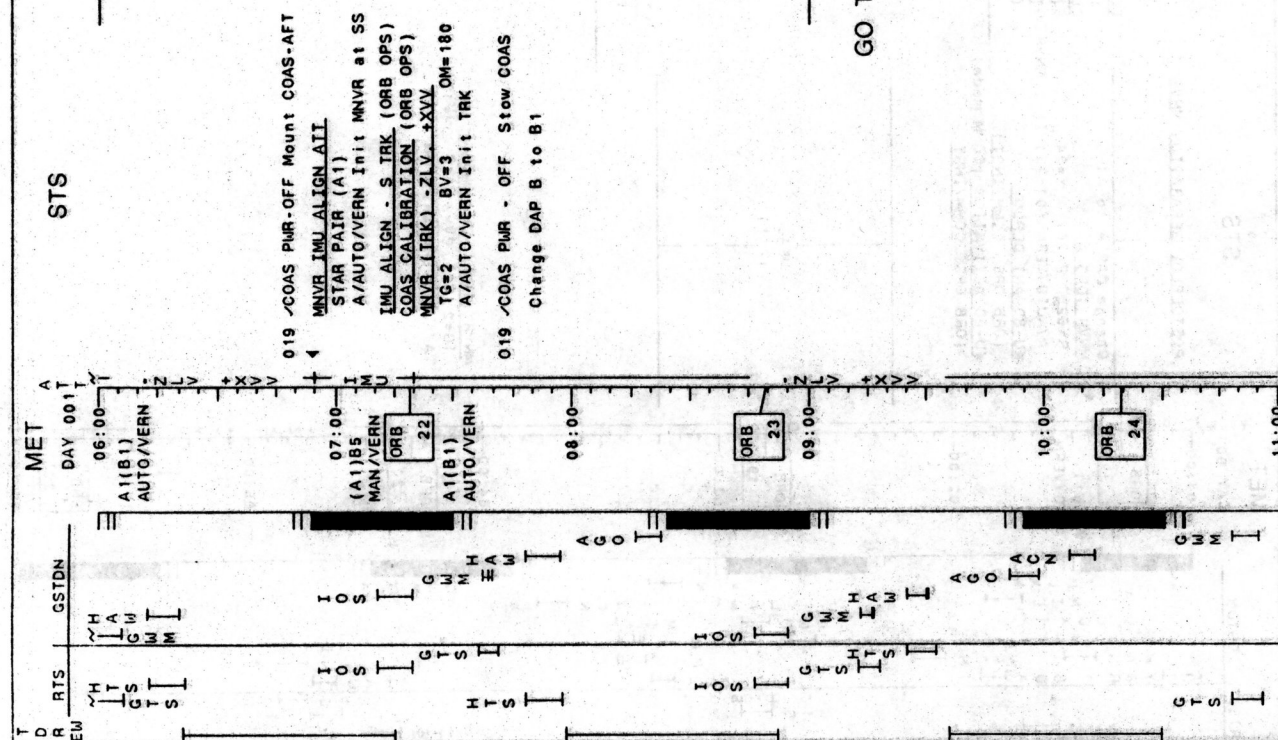
TG=2 BV=3 OM=180

A/AUTO/VERN Init TRK

STS 26 (FD02) 18A (EVA) INSERTION

7d

MCC



IMU ALIGN LOG
REQD ID:

-Y ____ : -Z ____ ANG ERR ____ . ____
 1 2 3

ANG:

$$\Delta^x \quad \Delta^y \quad \Delta^z \quad \Delta^w \quad \Delta^v \quad \Delta^u \quad \Delta^t \quad \Delta^s \quad \Delta^r \quad \Delta^q \quad \Delta^p \quad \Delta^o \quad \Delta^n \quad \Delta^m \quad \Delta^l \quad \Delta^k \quad \Delta^j \quad \Delta^i \quad \Delta^h \quad \Delta^g \quad \Delta^f \quad \Delta^e \quad \Delta^d \quad \Delta^c \quad \Delta^b \quad \Delta^a$$
 $\Delta^y \quad)_{-} \cdot - - ()_{-} \cdot - -$

31

EXECUTION TIME: () - () - ()

COAS CALIBRATION LOG

STAR ID: -2

STAR AVAIL: : TO :

-2 Δ BIAS — — — — —

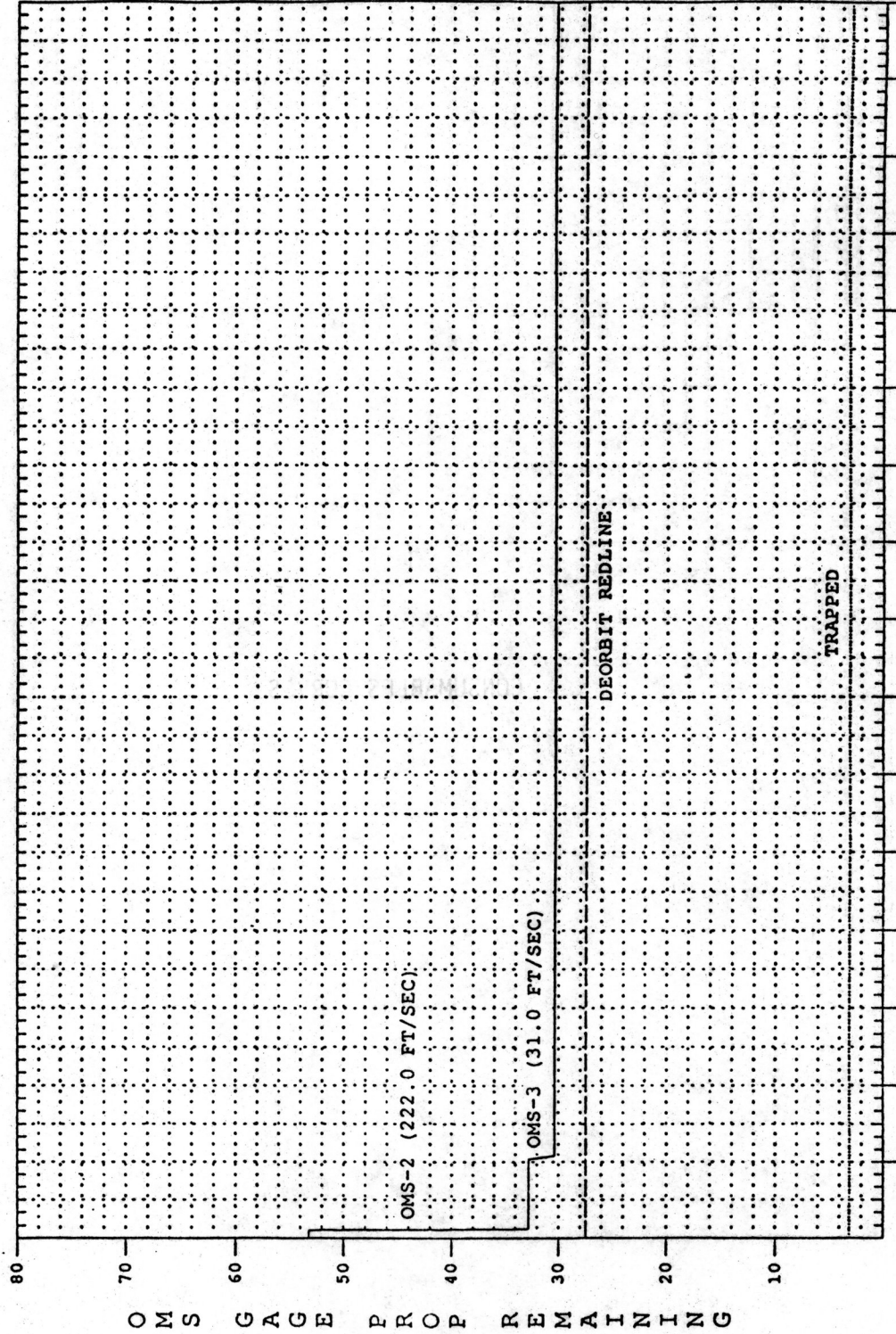
CONSUMABLES CURVES

CONSUMABLES
CURVES

CONSUMABLES CURVES

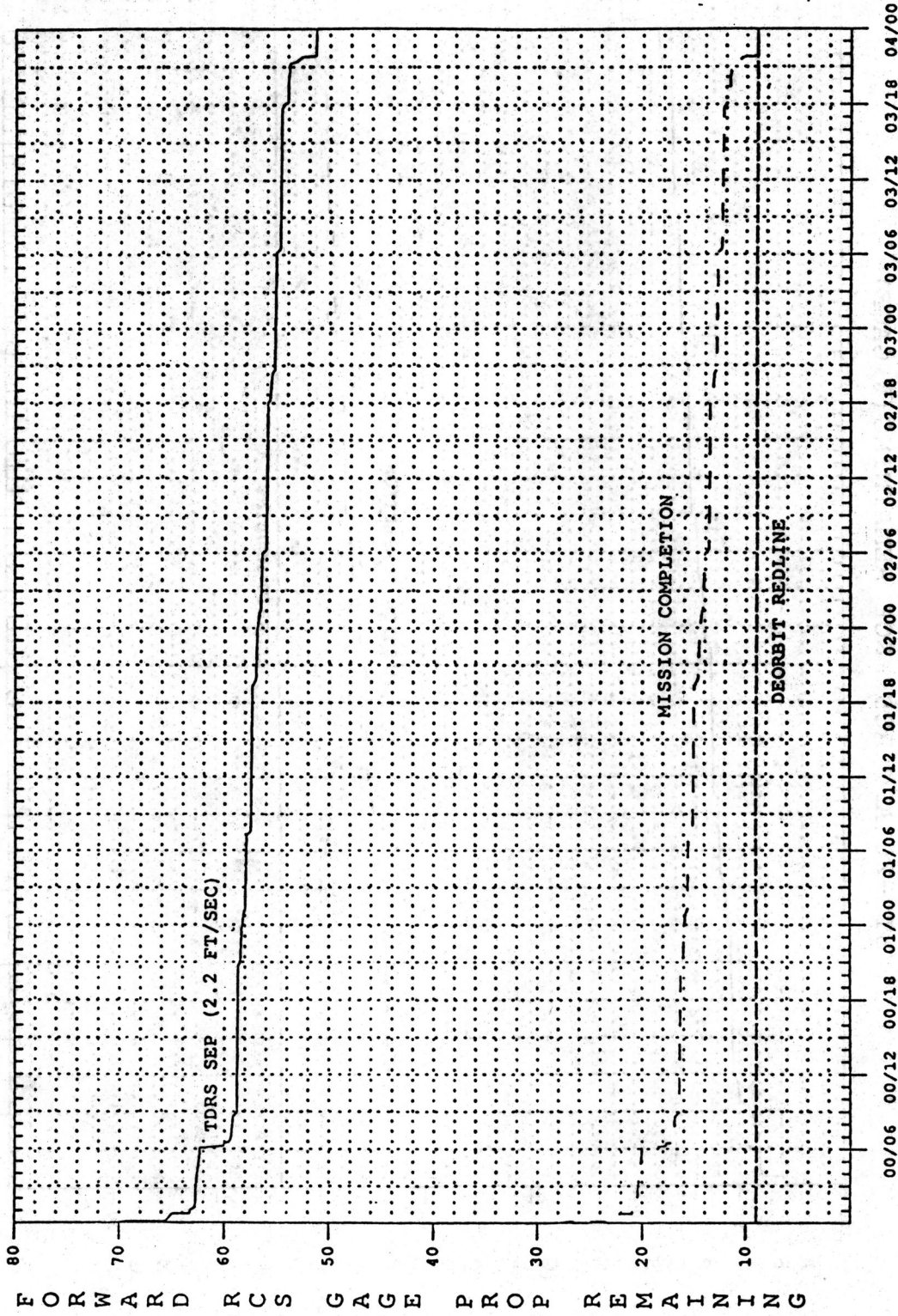
FDP: CY326B/May 88
CAP/ATL: Basic, 25 Nov. 87
Charts generated by: Bill Hollister/RSOC (DF75)
Charts generated on: 15 July, 88

STS-26 OMS PROP REMAINING (% GAGE)



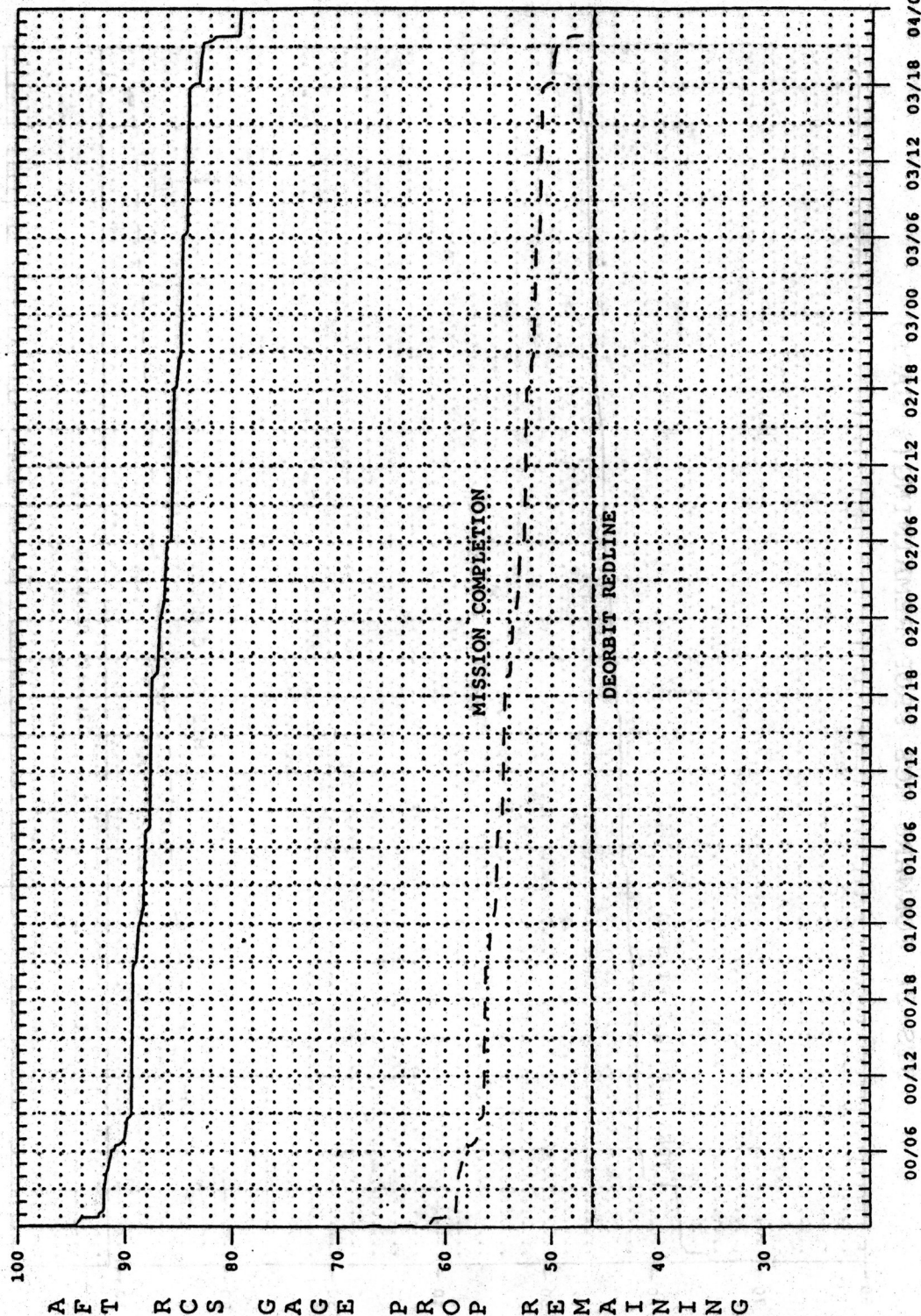
MISSION ELAPSED TIME IN DAYS/HOURS

STS-26 FORWARD RCS PROP REMAINING (% GAGE)



MISSION ELAPSED TIME IN DAYS/HOURS

STS-26 AFT RCS PROP REMAINING (% GAGE)



MISSION ELAPSED TIME IN DAYS/HOURS

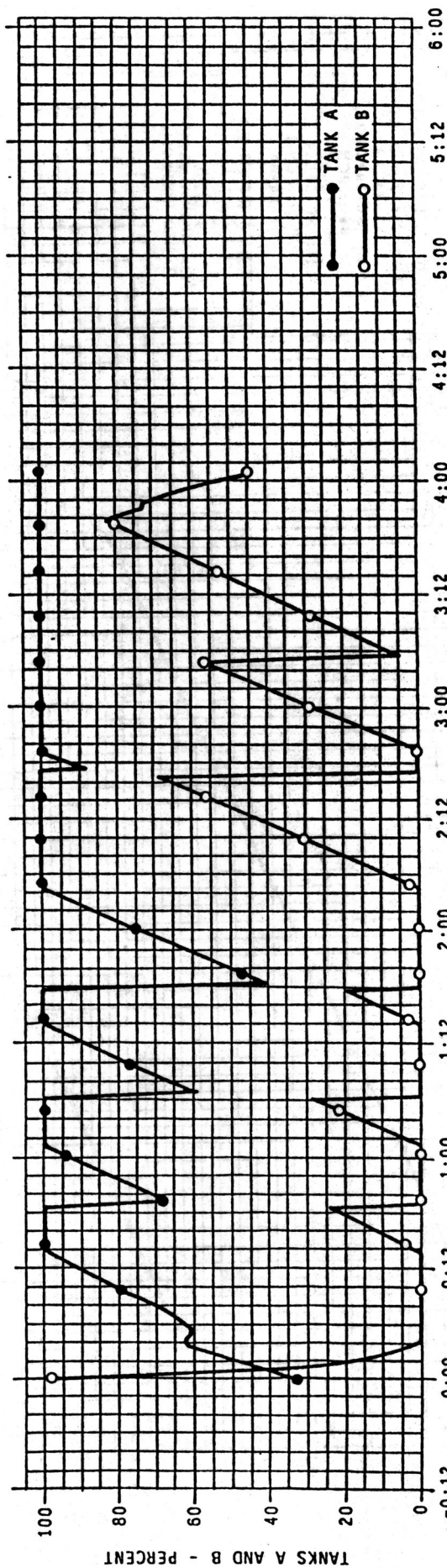
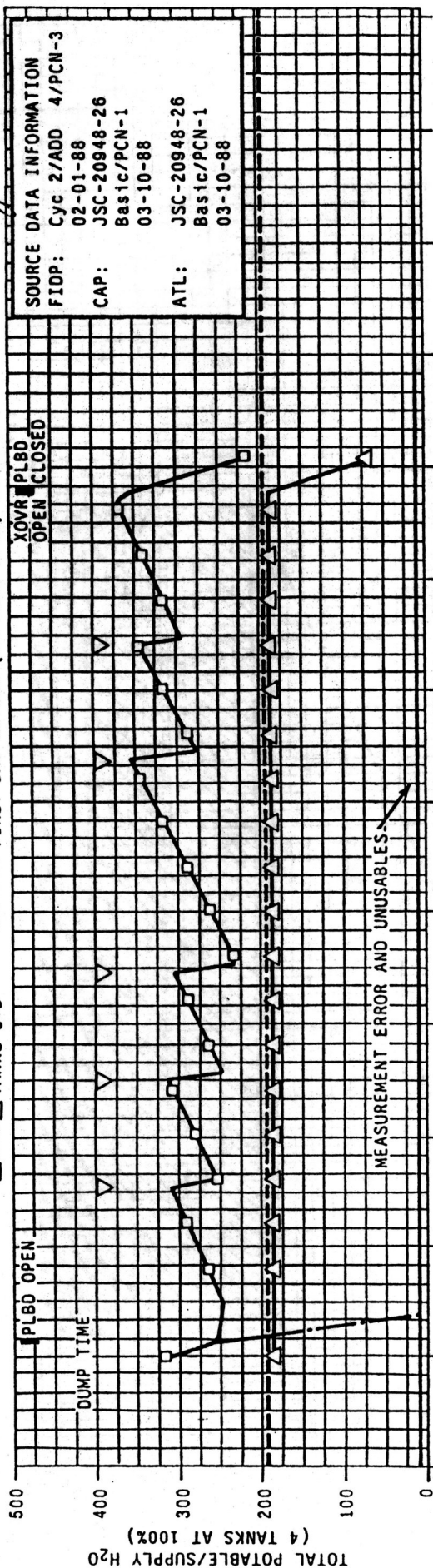
MISSION OPERATIONS DIRECTORATE-SYSTEMS DIVISION

KHANDAKER M. RAHMAN
ANALYSIS DATE: 04-04-88
TRAJ CYCLE: OFP-2

DATE CONFIRMED: *7/11/88*

--- ANYTIME CONTINGENCY DEORBIT
--- FIRST DAY RETURN (NO PLBD OPEN) REDLINE

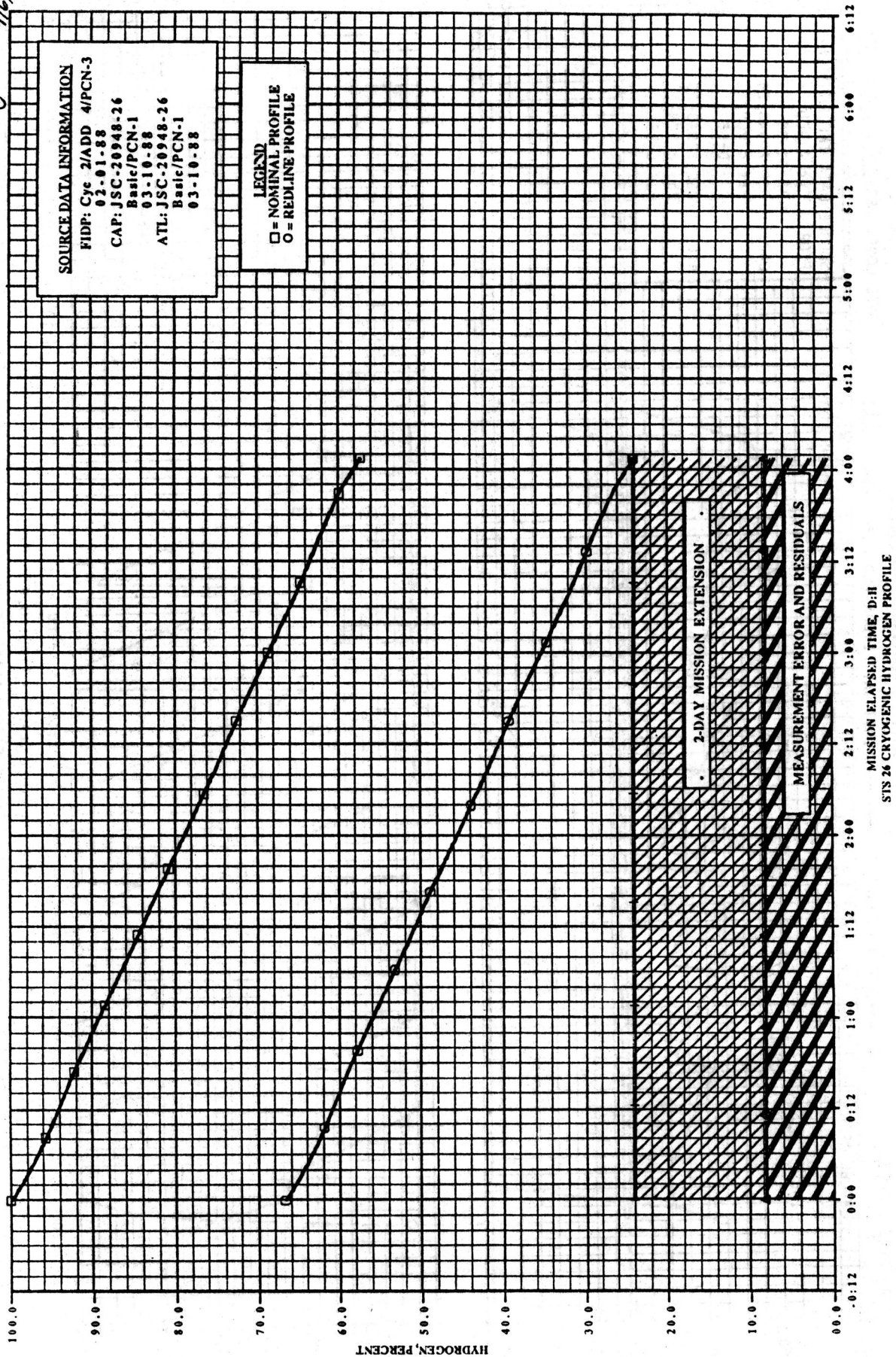
□ TANKS A-D
△ TANKS C+D



MISSIONS OPERATIONS DIRECTORATE
SYSTEMS DIVISION

R.D. PECHACEK
ANALYSIS DATE: 04-04-88
TRAJ CYCLE: OFP-2

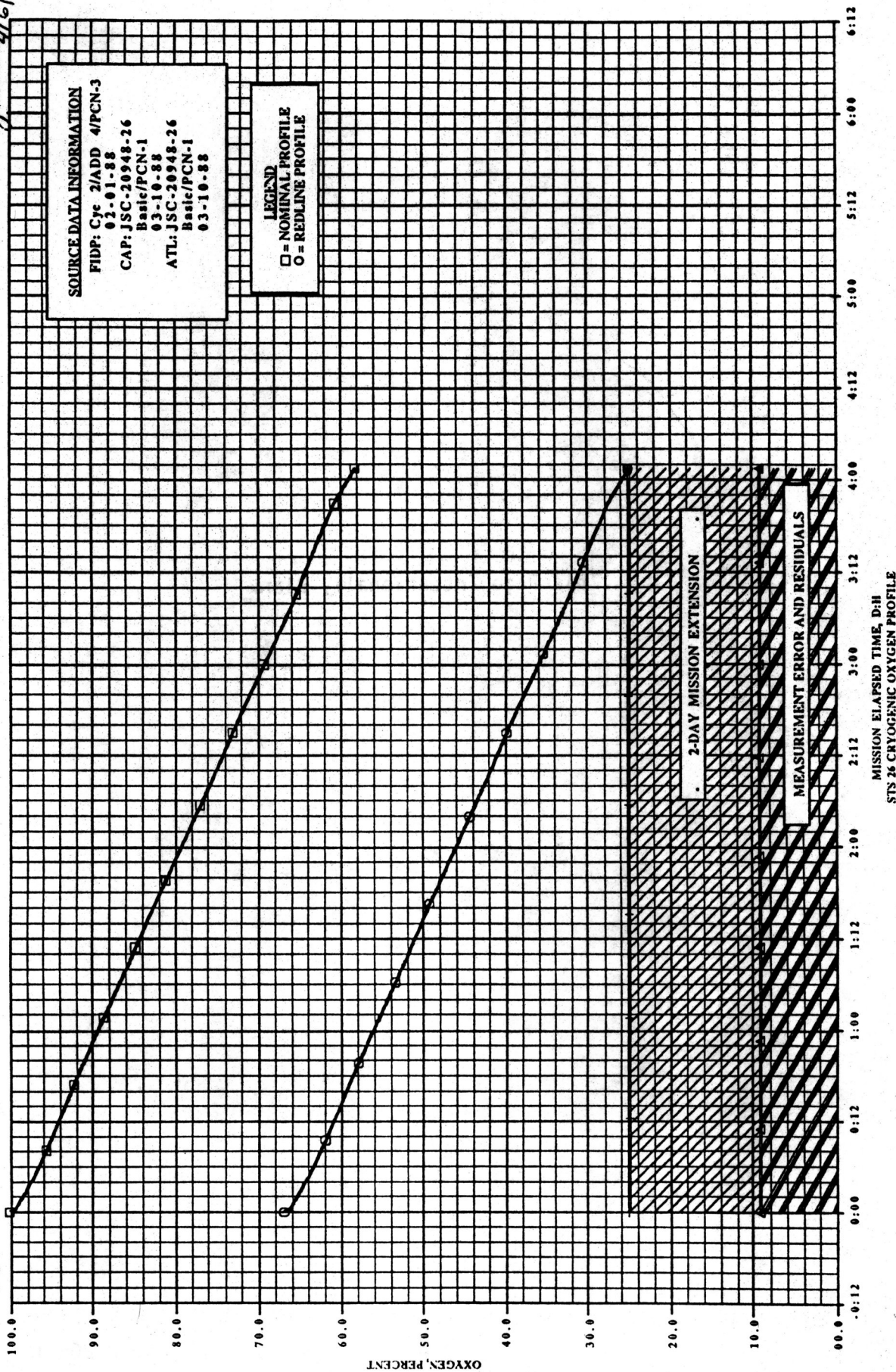
DATA CONFIRMED: *G. Clark* 7/6/88



MISSIONS OPERATIONS DIRECTORATE
SYSTEMS DIVISION

R. D. PECHACEK
ANALYSIS DATE: 04-04-88
TRAJ CYCLE: OFF-2

DATA CONFIRMED *J. Clark* 9/16/88



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The sections following this divider page
will not be included in flight copies.

The sections following this divider page
will not be included in flight copies.

CAP NOTES

A. FLIGHT DESCRIPTION

The STS-26 (OV-103) launch from KSC on October 4, 1988 (Cycle 3R) will be into a 160 nm orbit at an inclination of 28.5 degrees. The mission duration will be 4 days. Nominal landing occurs at EDW on October 8, 1988.

Landing opportunities are:

		<u>Landing</u>	<u>Site</u>	<u>Local Time</u>
		<u>Orbit No.</u>		
Nominal	Primary	65	EDW	08:00
	Backup	66	EDW	09:36
	Weather Alt.	65	NOR	09:03
1 Day Late	Primary	81	EDW	08:13
	Backup	82	EDW	09:48
	Weather Alt.	81	NOR	09:15
2 Days Late	Primary	96	EDW	06:49
	Backup	97	EDW	08:25
	Weather Alt.	96	NOR	07:52

MAJOR EVENTS

LAUNCH (KSC)	October 4, 1988
MET (DD/HH:MM:SS)	0/00:00:00
Day of year	278
CDT/GMT	09:07/14:03

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OMS-2
MET (DD/HH:MM:SS)      0/00:42:00
DV (fps)                 222
DT (MM:SS)               2:23
ORBIT ha/hp (nm)        159 x 161

```

TDRS/IUS DEPLOY (6A INSERTION)
MET (DD/HH:MM:SS) 0/06:14

<u>OMS SEPARATION BURN</u>	
MET (DD/HH:MM:SS)	0/06:29
DV (fps)	31
DT (MM:SS)	00:19
ORBIT ha/hp (nm)	177 x 161

DEORBIT BURN	
MET (DD/HH:MM:SS)	03/23:54
DV (fps)	304

LANDING (EDW)	
MET (DD/HH:MM:SS)	04/00:57:00
Day of year	282
CDT/GMT	10:00/15:00

B. CREW

1. The crewman responsible for performing an EVA, should one become necessary, are: MS3 will be EV1, MS1 will be EV2, and PLT will be IV1
2. There will be 5 crewmen flying this mission, with each specialist assigned to the following tasks:

CDR - Overall mission decisions, vehicle control, GNC, ECLSS, C&W, DPS, Medical
PLT - APU/HYD, MPS, EPS, OMS, RCS, Earth Obs
MS1 - IUS/TDRS, OASIS, Crew Equipment
MS2 - Communications, SPOC/HP41, PLBD/RAD
MS3 - EMU, TV, Instrumentation FDF, IFM Habitability, Middeck Payloads

C. ORBITER SYSTEMS

1. Communications and Instrumentation

- a. The Operational Instrumentation (OI) system will nominally be managed from the ground through uplink real-time commands coordinated with the crew.
- b. There are 13 GSTDN sites available for on-orbit coverage: CAN, GDX, MIL (MLX), BDA, HAW, GWM, AGO, ACN, MAD (MAX), YAR, DKR, IOS, and VAN
- c. There are 3 GSTDN sites available for real-time TV: GDX, MIL (MLX), and HAW.
- d. There are 2 GSTDN sites available for UHF voice only: YAR, IOS
- e. There are 4 SGLS sites available for on-orbit coverage during the TDRS/IUS mission: IOS, VTS, GTS, HTS.
- f. TDRS-E is assumed to be in position (41 degrees W).
- g. The only means available for HDR uplink (e.g., Teleprinter) while encrypted is through MIL, BDA and TDRS-E.

2. Guidance and Navigation

- a. The Orbiter state vector must be uplinked whenever downtrack error in the on-board vector becomes greater than 20 nm, but is usually uplinked when errors are much greater than 5 nm. These are real-time calls. In addition, state vectors must be uplinked for specified targets when utilizing an Orbiting Vehicle as a TGT ID with Universal Pointing.
- b. IMU alignments occur approximately every 12 hours in the morning and evening activities and prior to major events requiring low navigation uncertainty.
- c. In order not to perturb the Tracking Arc before the deploy, the first IMU align will be in a bias -ZLV $\pm$ YVV attitude.

The orbiter will be biased in yaw assuring acquisition of stars for the IMU align before the IUS deploy.

3. Environmental Control and Life Support Subsystem (ECLSS)

- a. The CO2 absorbers are installed for launch and entry and are alternately replaced on-orbit as required.
- b. There are four supply water tanks and one waste water tank which are loaded for launch with volumes sufficient to meet an orbit 3 launch day deorbit with PLBDs failed closed. If waste water production exceeds storage capacity, a real-time call will be made to perform a waste tank dump. Normally two supply water dumps will occur per flight day. A real-time call will be made prior to each to inform the crew whether a dump is required, and to what level. Tanks A and B will be dumped to a level that will allow the tanks to be full prior to the next daily group of deorbit opportunities.

4. DC Utility Power Usage

The following table is the DC Utility Power Outlet Schedule. This table shows the times of outlet reconfigurations during the mission.

STS-26 DC UTILITY POWER OUTLET SCHEDULE

MET OUTLET	PRELAUNCH	POST INSERTION	ON ORBIT				DEORBIT PREP
		0/02:00	0/08:45	0/09:00	3/19:00	3/23:00	
019	PLT*	----	----	----	----	----	PLT*
MO52J	MS3*	----	----	ADSF	----	----	MS3*
F1	----	----	----	----	----	----	----
MO13Q	PDB**	----	PVTOS	PVTOS	PDB**	----	PDB**
A11	MS1*, MS2*	----	----	----	----	----	MS1*, MS2*
A15	CDR*	TLPR	TLPR	TLPR	TLPR	----	CDR*
MO30F	----	PDB**	PDB**	PDB**	----	----	----

NOTES:

* Crewman suit fan

** Power distribution box with ARC and PCG

If an outlet column is blank, the outlet is not in use

D. TEXT UPLINK SYSTEM

1. The teleprinter (TPR) will provide an on-orbit capability to receive and reproduce text-only data (such as procedures, weather data, and CAP updates or changes) from MCC. The TPR uses S-Band and is not dependent on TDRS Ku-Band. TPR uplink requires 1 to 2.5 minutes per message, depending upon the number of lines (up to 66). Activation includes powering up the unit during Post Insertion. The TPR will remain on continuously until deactivation prior to seat ingress on entry day.

E. TELEVISION/PHOTOGRAPHY

1. The CCTV systems provide two color wide-angle cameras for in-cabin coverage. There will also be four PLBD TV cameras and two Remote Manipulator System (RMS) TV cameras (RMS Elbow, RMS end effector). The two RMS TV cameras will only be available if the RMS is carried on the flight. There are three different lens assemblies for the PLB and RMS TV cameras (color lens assembly, monochrome lens assembly and wide angle lens assembly). The type of lens assembly for each PLB and RMS TV camera can be selected pre-flight by the payload customer.
2. There is a video tape recorder available with 30 min cassettes.
3. 35mm, 16mm, and 70mm camera systems are available.

F. PAYLOADS

1. TDRS (Tracking and Data Relay Satellite)

The TDRS-C is a communication satellite which is the second of two spacecraft launched in the process of assembling the Tracking and Data Relay Satellite System. The system will provide a high-capacity communication and data link with the STS as well as other spacecraft and launch vehicles. TDRS-A was launched on STS-6. The nominal IUS/TDRS-C deployment opportunity occurs 60 minutes prior to node 6A (ascending) at 0/6:14 MET. Backup opportunities occur prior to 7A and 16A. A contingency opportunity occurs 60 minutes prior to 8D (descending).

2. IUS (Inertial Upper Stage)

The IUS is a two-stage solid rocket, inertially stabilized upper stage which will place TDRS-C in a geosynchronous orbit. The IUS ignites its first stage (SRM-1) for transfer orbit insertion 60 minutes after TDRS-C/IUS deployment. The second burn (SRM-2) occurs approximately 7 hours after deployment, resulting in a 0 degree inclination, geosynchronous orbit at 178 degree W longitude for the 6A deployment opportunity.

3. OASIS-I (Orbiter Experiments Autonomous Supporting Instrumentation System)

The OASIS-I payload will collect environmental data for Orbiter primary payloads. OASIS is capable of collecting a variety of environmental parameters based on specific transducer configurations for each flight. For STS-26, OASIS is configured to interface with TDRS-C/IUS. OASIS has transducers/sensors mounted on the forward and aft IUS ASE to measure thermal, acoustic, vibration, stress, and acceleration data.

4. ADSF (Automated Directional Solidification Furnace)

The ADSF is a furnace facility designed to conduct materials processing experiments. It takes the space of 5 middeck lockers. It consists of a furnace container which houses 4 samples, a Control Electronics Container (CEC), and a Middeck Electronics Module (MEM). The 4 samples will run for 10 hours 30 minutes each.

5. PVTOS (Physical Vapor Transport of Organic Solids)

PVTOS is an experiment which will observe and study sublimation of 9 organic solids in a micro-g environment via gaseous diffusion. It takes the space of 3 Middeck lockers and consists of a Experiment Apparatus Container (EAC), and a Generic Electronics Module (GEM). Each of the 9 cells takes 5 hours of processing.

6. ARC (Aggregation of Red Blood Cells)

The ARC payload will conduct experiments which will supply information on the kinetics of aggregation of human red blood cells, the morphology of red blood cell aggregates, and the viscosity of whole blood at lower and higher flow rates. It consists of 2 sealed packages, the Experiment Apparatus Container (EAC) and the Electronics Equipment Package (EEP). It takes the place of 3 middeck lockers. The experiment run time is 8 hours.

7. IEF (Isoelectric Focusing Experiment)

IEF is a payload which will evaluate the effect of electro-osmosis on isoelectric focusing in the absence of gravity. The IEF is a completely self-contained assembly which takes up 1 Middeck locker. Experiment duration is 1.5 hours.

8. MLE (Mesoscale Lightning Experiment)

The objective of MLE is to photograph large scale upper atmospheric lightning using the Cargo Bay cameras and a 35 mm camera. It is a target of opportunity observation where the crew will document lightning phenomenon at one of 6 designated areas. It requires 2 crewmen to make the observation. Observations are to be made on at least 2 night passes per flight day.

9. ELRAD (Earth Limb Radiance Experiment)

The ELRAD payload will measure the radiance of twilight sky for various positions of the sun from sunset (and sunrise) to 9 degrees below the earth's horizon. It requires 1 crewman to perform the experiment, a 35 mm camera with an 85 mm lens and 3 photographic sequences (1 sunrise and 2 sunsets).

10. IRCFE (Infrared Communication Flight Experiment)

The purpose of this payload is to demonstrate the use of infrared communication. This system will replace the radio frequency communication system used by 1 crewman for a 2 hour period. Operation should occur during the passage through the South Atlantic Anomaly.

11. PPE (Phased Partitioning Experiment)

The PPE payload will document the phenomena of surface tension effects on mixed fluids in the absence of gravity. The experiment consists of a module with 15 chambers of mixed fluids. One crewman will shake up the module and document the separation of the fluid at intervals of time.

12. SSIP 82-5, 82-4 (Shuttle Student Involvement Projects)

Student experiment 82-4 is called "The Effect of Weightlessness on Grain Formation and Strength in Metals". Its objective is to observe the transformation and resolidification of Titanium alloys in a zero-g environment. It utilizes 1/2 a middeck locker. It contains 4 Titanium alloy samples contained in pyrex mounted on the experiment base plate. One crewman will send a 10 second charge through each of the wires and they will be analyzed after the mission.

Student experiment 82-5 is called "Utilizing a Semi-Permeable Membrane to Direct Crystal Growth in Zero-Gravity". Its objective is to observe and obtain crystal growth between 2 aqueous solutions in a deionized water solution. The experiment uses 1/2 middeck locker. It contains 3 crystal growth units. The unit will be activated by turning the handles on each end of the units. The Lead Acetate solution and the Potassium Iodide solution will mix in the deionized water and promote crystal growth. The crystal growth will be documented on video and analyzed post mission.

13. PCG (Protein Crystal Growth)

The objective of the PCG is to produce large, high quality crystals of selected proteins under controlled conditions in microgravity. The payload mounts in place of 1 middeck locker. Experiment operation consists of initiation of the crystal growth once on orbit, and photography at 24 hours intervals.

G. ON-ORBIT DSO'S

1. Salivary Pharmacokinetics of Scop-Dex (DSO 457)

The purpose of this DSO is to investigate the pharmacokinetics of scopolamine and dextroamphetamine during space flight. When the degree and nature of the pharmacokinetics are determined, the therapeutic consequences may be predicted. The crewmember will take the drug after an 8 hour fast, and take salivary samples at required intervals during the flight day.

2. Salivary Acetaminophen Pharmacokinetics (DSO 458)

This DSO investigates the pharmacokinetics of Acetaminophen (Tylenol). This drug is used for evaluation because it distributes into the saliva with a ratio similar to blood plasma. The crewmembers will take the Tylenol after an 8 hour fast and take salivary samples at specified intervals during the flight day.

3. Otolith Tilt-Translation Reinterpretation (DSO 459)

The objective of this DSO is to collect data which will allow development of a trainer to preadapt astronauts to the sensory changes produced by microgravity. Two crewmembers will work together to obtain data regarding their perceived position each flight day. One crewmember will wear the OTTR goggles for data collection of otolith response.

4. Changes in Total Body Water During Spaceflight (DSO 460)

The purpose of this DSO is to measure the changes in total body water that occur during spaceflight. These measurements will also prove valuable in future studies which will track fluid shifts among specific body fluid compartments. The procedure involves ingestion of a dose of water labeled with the stable isotope oxygen-18, followed by periodic collection of saliva samples to determine the concentration of labeled water after establishment of a steady-state distribution.

5. Documentary Television (DSO 901)

This DSO requires live television transmission or VTR dumps of crew activities and spacecraft functions which include: Payload Bay Views, TDRS-C/IUS Predeploy and Deploy Activities, VTR Downlink of Deploy Activities, In-Flight Crew Conference, and unscheduled TV Activities.

6. Documentary Motion Picture Photography (DSO 902)

This DSO requires documentary and public affair motion picture photography of significant activities which best depicts the basic capabilities of the Space Shuttle and key objectives. This DSO includes motion picture photography of TDRS-C/IUS Predeploy and Deploy Activities, Flight Deck Activities, Middeck Activities, and any unscheduled Motion Picture Photography.

7. Documentary Still Photography (DSO 903)

This DSO requires still photography of crew activities in the orbiter, spacecraft accommodations and mission related scenes of general public and historical interest. Still photography with 70mm format for exterior photography and 35mm format for interior photography is required.

H. ON-ORBIT DTOS

1. Cabin Air Monitoring (DTO 623)

This DTO will use the solid sorbent sampler to continuously sample the Orbiter atmosphere throughout the flight. The solid sorbent sampler is to be flown on first flights of a new vehicle or after extensive non-metallic materials refurbishment in the Orbiter cabin.

2. Attitude Match Update (DTO 787)

Downlink Orbiter and IUS data will be compared at 6 points during the predeploy timeline. CSTC will receive the data and test and verify the new procedures for evaluating and correcting alignment errors onboard the IUS.

I. TIMELINE FORMATS

1. The timeline formats used in this Crew Activity Plan are based on JSC document 19933: Timeline Format Definitions and Standard Notes, Revision B, May 25, 1987.

NOTATION									
1	2	3	4	5	6	7	8	9	10
11	12	13	14	15	16	17	18	19	20
21	22	23	24	25	26	27	28	29	30
31	32	33	34	35	36	37	38	39	40
41	42	43	44	45	46	47	48	49	50
51	52	53	54	55	56	57	58	59	60
61	62	63	64	65	66	67	68	69	70
71	72	73	74	75	76	77	78	79	80
81	82	83	84	85	86	87	88	89	90
91	92	93	94	95	96	97	98	99	100

STS-26 DAP BOOK

DAP A CONFIGURATIONS

	A1*	A2	A3	A4	A5	A6	A7	A8	A9	A10	A11
<u>TRANSLATION</u>											
PULSE ft/sec	0.10	0.10	0.10	0.10	0.10	0.10		0.10	0.10	0.05	0.25
<u>ROTATION</u>											
DSC RT NORM °/sec	0.200	0.400	0.200	0.200	0.200	0.200		0.500	0.200	0.200	0.200
VERN °/sec	0.200	0.400	0.200	0.008	0.200	0.200		0.200	0.200	0.200	0.200
PULSE NORM °/sec	0.10	0.10	0.30	0.10	0.10	0.10		0.10	0.10	0.10	0.05
VERN °/sec	0.010	0.010	0.010	0.010	0.050	0.010		0.010	0.010	0.010	0.010
COMP NORM °/sec	.00	.00	.00	.00	.00	.00		.00	.00	.00	.00
VERN °/sec	.000	.000	.000	.000	.000	.000		.000	.000	.000	.000
<u>DEADBAND</u>											
ATT NORM (deg)	5.00	1.00	5.00	1.00	5.00	5.00		3.00	2.00	2.00	1.00
VERN (deg)	1.000	1.000	1.000	0.070	1.000	1.000		1.000	1.000	1.000	1.000
RATE NORM °/sec	0.20	0.20	0.20	0.20	0.20	0.20		0.20	0.20	0.20	0.20
VERN °/sec	.020	.020	.020	.010	.020	.020		.020	.020	.020	.010
<u>JET OPT</u>											
P	1	1	3	1	1	3		1	1	1	1
Y	1	1	3	3	1	3		1	1	1	1
CNTL ACCEL	0	0	0	0	0	0		0	0	0	0

*I-LOADED OPS-2 DAP Configuration

DAP B CONFIGURATIONS

	B1*	B2	B3	B4	B5	B6	B7	B8
<u>TRANSLATION</u>								
PULSE ft/sec	0.02	0.02	0.02	0.02	0.02	0.05	0.01	
<u>ROTATION</u>								
DSC RT NORM °/sec	0.500	0.200	0.200	0.200	0.200	0.500	0.200	
VERN °/sec	0.200	0.200	0.200	0.008	0.016	0.200	0.200	
PULSE NORM °/sec	0.04	0.04	0.04	0.04	0.04	0.04	0.04	
VERN °/sec	0.002	0.002	0.002	0.002	0.002	0.002	0.002	
COMP NORM °/sec	.00	.00	.00	.00	.00	.00	.00	
VERN °/sec	.000	.000	.000	.000	.000	.000	.000	
<u>DEADBAND</u>								
ATT NORM (deg)	3.00	3.00	3.00	0.30	3.00	2.00	2.00	
VERN (deg)	1.000	1.000	1.000	0.100	0.033	1.000	1.000	
RATE NORM °/sec	0.20	0.20	0.20	0.20	0.20	0.20	0.20	
VERN °/sec	.020	.020	.020	.010	.020	.020	.020	
<u>JET OPT</u>								
P	1	3	1	3	3	1	1	
Y	1	3	1	3	3	1	1	
<u>CNTL ACCEL</u>	0	0	0	0	0	0	0	

*I-LOADED OPS-2 DAP Configuration

DAP BOOK

HOOK
VELCRO

CAP/26/0/BAS

CAP/26/0/BAS

UNIV PTG	
TGT ID = 1	Orbiting Vehicle
= 2	Earth ctr
= 3	Earth tgt
	Input LAT (± 90)
	LON (± 180 , + = East)
	ALT (-3444.0 to 20000.0 nm)
= 4	Sun ctr
= 5	Celestial tgt
	Input RA (0-359.99)
	DEC (± 90)
= 11-110	Nav Stars
BODY VECT = 1	+X*
= 2	-X*
= 3	-Z*
= 4	PTC ($\sim 2^\circ$ below +X)*
= 5	Optional
	Input P (0-359.99)
	Y (270-359.99, 0-90)
* input OM ->	OM (0-359.99)
MON AXIS = 1	+X
2	-X

G20 DAP CONFIG	
CNTL ACCEL = 0	Orbiter alone
= 1	thru 5 not used
JET OPT = 1	Normal
= 2	Nose only
= 3	Tail only

TOP

DAP PURPOSE

A1 - NOMINAL
IMU ALIGN
LOSS OF VERN
(NOSE & TAIL)
A2 - PTC
A3 - RCS HOT FIRE
A4 - GG
A5 - IMU ROTATION
A6 - LOSS OF VERN
(TAIL ONLY)
A7 - (Blank)
A8 - PRE/POST DEPLOY
A9 - RENDEZVOUS
A10 - PROX OPS
A11 - IUS DEPLOY

CAP/26/0/BAS

A1

TOP

DAP A1

PURPOSE:

NOMINAL
IMU ALIGN
LOSS OF VERN
(NOSE & TAIL)

A1

0.1 0	1	TRANSLATION PULSE
0.2 0 0	2	ROTATION
0.2 0 0	3	DSC RT NORM
0.1 0 0	4	VERN
0.0 1 0	5	PULSE NORM
.0 0 0	6	VERN
.0 0 0	7	COMP NORM
		VERN
5.0 0	8	DEADBAND
1.0 0 0	9	ATT NORM
		VERN
0.2 0	10	RATE NORM
.0 2 0	11	VERN
1	12	JET OPT P
1	13	Y
0	14	CNTL ACCEL

CAP/26/0/BAS

CAP/26/0/BAS

A2

TOP

(reduced copy)

TOP

A1

DAP A2

PURPOSE:

PTC

A2

0.1 0

1 TRANSLATION
PULSE

0.4 0 0

2 ROTATION

DSC RT NORM

0.4 0 0

3 VERN

0.1 0 0

4 PULSE NORM

0.0 1 0

5 VERN

0.0 0 0

6 COMP NORM

0.0 0 0

7 VERN

DEADBAND

1.0 0 0

8 ATT NORM

1.0 0 0

9 VERN

0.2 0 0

10 RATE NORM

0.2 0 0

11 VERN

1

12 JET OPT P

1

13 Y

0

14 CNTL ACCEL

CAP/26/0/BAS

CAP/26/0/BAS

A3

TOP

TOP

A2

DAP A3

PURPOSE:

RCS HOT FIRE

0.1 0

1 TRANSLATION
PULSE

0.2 0 0

2 ROTATION

DSC RT NORM

0.2 0 0

3 VERN

0.3 0 0

4 PULSE NORM

0.0 1 0

5 VERN

0.0 0 0

6 COMP NORM

0.0 0 0

7 VERN

DEADBAND

5.0 0 0

8 ATT NORM

1.0 0 0

9 VERN

0.2 0 0

10 RATE NORM

0.2 0 0

11 VERN

3

12 JET OPT P

3

13 Y

0

14 CNTL ACCEL

CAP/26/0/BAS

CAP/26/0/BAS

A4

TOP

(reduced copy)

7-6

CAP/26/FIN C

TOP

A3

DAP A4

PURPOSE:

GG

A4

0.1 0

1

TRANSLATION
PULSE

0.2 0 0

2

ROTATION
DSC RT NORM

0.0 0 0

3

VERN

0.1 0 0

4

PULSE NORM

0.0 1 0

5

VERN

.0 0 0

6

COMP NORM

.0 0 0

7

VERN

DEADBAND

1.0 0 0

8

ATT NORM

0.0 7 0

9

VERN

0.2 0 0

10

RATE NORM

.0 1 0

11

VERN

1

12

JET OPT P

3

13

Y

0

14

CNTL ACCEL

CAP/26/0/BAS

CAP/26/0/BAS

A5

TOP

TOP

A4

DAP A5

PURPOSE:

IMU ROTATION

A5

0.1 0

1

TRANSLATION
PULSE

0.2 0 0

2

ROTATION
DSC RT NORM

0.2 0 0

3

VERN

0.1 0 0

4

PULSE NORM

0.0 5 0

5

VERN

.0 0 0

6

COMP NORM

.0 0 0

7

VERN

DEADBAND

5.0 0 0

8

ATT NORM

1.0 0 0

9

VERN

0.2 0 0

10

RATE NORM

.0 2 0

11

VERN

1

12

JET OPT P

1

13

Y

0

14

CNTL ACCEL

CAP/26/0/BAS

CAP/26/0/BAS

A6

TOP

(reduced copy)

7-7

CAP/26/FIN C

TOP

A5

DAP A6

PURPOSE:

LOSS OF VERN
(TAIL ONLY)

A6

0.1 0 1 TRANSLATION
PULSE

0.2 0 0	2	DSC RT	NORM
0.2 0 0	3		VERN
0.1 0 0	4	PULSE	NORM
0.0 1 0	5		VERN
.0 0 0	6	COMP	NORM
.0 0 0	7		VERN

5.0 0 0	8	DEADBAND	
1.0 0 0	9	ATT	NORM
			VERN

0.2 0 0	10	RATE	NORM
.0 2 0	11		VERN

3	12	JET OPT	P
3	13		Y

0 14 CNTL ACCEL

CAP/26/0/BAS

CAP/26/0/BAS

A7

TOP

TOP

A6

DAP A7

PURPOSE: (Blank)

A7

-.- 1 TRANSLATION
PULSE

-.-	2	DSC RT	NORM
-.-	3		VERN
-.-	4	PULSE	NORM
-.-	5		VERN
-.-	6	COMP	NORM
-.-	7		VERN

-.-	8	DEADBAND	
-.-	9	ATT	NORM
			VERN

-.-	10	RATE	NORM
-.-	11		VERN

-	12	JET OPT	P
-	13		Y

- 14 CNTL ACCEL

CAP/26/0/BAS

CAP/26/0/BAS

A8

TOP

(reduced copy)

7-8

CAP/26/FIN C

TOP

A7

DAP A8

PURPOSE:

PRE/POST DEPLOY

A8

0.1 0	1	TRANSLATION	
		PULSE	

ROTATION			
0.5 0 0	2	DSC RT	NORM
0.2 0 0	3		VERN
0.1 0 0	4	PULSE	NORM
0.0 1 0	5		VERN
0.0 0 0	6	COMP	NORM
0.0 0 0	7		VERN

DEADBAND			
3.0 0	8	ATT	NORM
1.0 0 0	9		VERN

0.2 0	10	RATE	NORM
0.0 2 0	11		VERN

1	12	JET OPT	P
1	13		Y

0	14	CNTL ACCEL	
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CAP/26/0/BAS

CAP/26/0/BAS

A9

TOP

TOP

A8

DAP A9

PURPOSE:

RENDEZVOUS

A9

0.1 0	1	TRANSLATION	
		PULSE	

ROTATION			
0.2 0 0	2	DSC RT	NORM
0.2 0 0	3		VERN
0.1 0 0	4	PULSE	NORM
0.0 1 0	5		VERN
0.0 0 0	6	COMP	NORM
0.0 0 0	7		VERN

DEADBAND			
2.0 0	8	ATT	NORM
1.0 0 0	9		VERN

0.2 0	10	RATE	NORM
0.0 2 0	11		VERN

1	12	JET OPT	P
1	13		Y

0	14	CNTL ACCEL	
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CAP/26/0/BAS

CAP/26/0/BAS

A10

TOP

(reduced copy)

CAP/26/FIN C

TOP

A9

DAP A10

PURPOSE:

PROX OPS

A10

TRANSLATION
PULSE

0.0 5

ROTATION

0.2 0 0	2	DSC RT	NORM
0.2 0 0	3	VERN	
0.1 0 0	4	PULSE	NORM
0.0 1 0	5	VERN	
0.0 0 0	6	COMP	NORM
0.0 0 0	7	VERN	

DEADBAND

2.0 0	8	ATT	NORM
1.0 0 0	9		VERN

0.2 0	10	RATE	NORM
0.2 0	11		VERN

1	12	JET OPT	P
1	13		Y

0 14 CNTL ACCEL

CAP/26/0/BAS

CAP/26/0/BAS

A11

TOP

TOP

A10

DAP A11

PURPOSE:

IUS DEPLOY

TRANSLATION
PULSE

0.2 5

ROTATION

0.2 0 0	2	DSC RT	NORM
0.2 0 0	3	VERN	
0.0 5	4	PULSE	NORM
0.0 1 0	5	VERN	
0.0 0 0	6	COMP	NORM
0.0 0 0	7	VERN	

DEADBAND

1.0 0	8	ATT	NORM
1.0 0 0	9		VERN

0.2 0	10	RATE	NORM
0.1 0	11		VERN

1	12	JET OPT	P
1	13		Y

0 14 CNTL ACCEL

CAP/26/0/BAS

CAP/26/0/BAS

TOP

(reduced copy)

7-10

CAP/26/FIN C

TOP

A11

DAP PURPOSE

- B1 - NOMINAL
PRCS IMU ALIGN
OMS & RCS BURNS
- B2 - LOSS OF VERN
(TAIL ONLY)
- B3 - LOSS OF VERN
(NOSE & TAIL)
GG MNVR
- B4 - SSUS DEPLOY
- B5 - COAS CAL
- B6 - RENDEZVOUS
- B7 - PROX OPS
- B8 - (Blank)

CAP/26/O/BAS

CAP/26/O/BAS

B1

TOP

TOP

DAP B1

B1

PURPOSE:

NOMINAL
PRCS IMU ALIGN
OMS & RCS BURNS

TRANSLATION

PULSE 15 0.0 2

ROTATION

DSC RT NORM 16 0.5 0 0

VERN 17 0.2 0 0

PULSE NORM 18 0.0 4

VERN 19 0.0 0 2

COMP NORM 20 0.0 0

VERN 21 0.0 0 0

DEADBAND

ATT NORM 22 3.0 0

VERN 23 1.0 0 0

RATE NORM 24 0.2 0

VERN 25 0.0 2 0

JET OPT P 26 1

Y 27 1

CNTL ACCEL 28 0

CAP/26/O/BAS

CAP/26/O/BAS

B2

TOP

(reduced copy)

TOP

B1

DAP B2

PURPOSE:

LOSS OF VERN
(TAIL ONLY)

B2

TRANSLATION
PULSE

15 0.0 2

ROTATION

DSC RT	NORM	16	0.2	0	0
	VERN	17	0.2	0	0
PULSE	NORM	18	0.0	4	
	VERN	19	0.0	0	2
COMP	NORM	20	0.0	0	
	VERN	21	0.0	0	0

DEADBAND

ATT	NORM	22	3.0	0	
	VERN	23	1.0	0	0

RATE	NORM	24	0.2	0	
	VERN	25	0.2	0	0

JET OPT	P	26	3		
	Y	27	3		

CNTL ACCEL 28 0

CAP/26/0/BAS

CAP/26/0/BAS

B3

TOP

TOP

B2

DAP B3

PURPOSE:

LOSS OF VERN
(NOSE & TAIL)
GG MNVR

B3

TRANSLATION
PULSE

15 0.0 2

ROTATION

DSC RT	NORM	16	0.2	0	0
	VERN	17	0.2	0	0
PULSE	NORM	18	0.0	4	
	VERN	19	0.0	0	2
COMP	NORM	20	0.0	0	
	VERN	21	0.0	0	0

DEADBAND

ATT	NORM	22	3.0	0	
	VERN	23	1.0	0	0

RATE	NORM	24	0.2	0	
	VERN	25	0.2	0	0

JET OPT	P	26	1		
	Y	27	1		

CNTL ACCEL 28 0

CAP/26/0/BAS

CAP/26/0/BAS

B4

TOP

(reduced copy)

7-12

CAP/26/FIN C

TOP

B3

DAP B4

PURPOSE:

SSUS DEPLOY

B4

TRANSLATION
PULSE 15 0.0 2

ROTATION			
DSC RT	NORM	16	0.2 0 0
	VERN	17	0.0 0 8
PULSE	NORM	18	0.0 4
	VERN	19	0.0 0 2
COMP	NORM	20	.0 0 0
	VERN	21	.0 0 0

DEADBAND			
ATT	NORM	22	0.3 0
	VERN	23	0.1 0 0

RATE	NORM	24	0.2 0
	VERN	25	.0 1 0

JET OPT	P	26	3
	Y	27	3

CNTL ACCEL 28 0

CAP/26/0/BAS

CAP/26/0/BAS

B5

TOP

TOP

B4

DAP B5

PURPOSE:

COAS CAL

B5

TRANSLATION
PULSE 15 0.0 2

ROTATION			
DSC RT	NORM	16	0.2 0 0
	VERN	17	0.0 1 6
PULSE	NORM	18	0.0 4
	VERN	19	0.0 0 2
COMP	NORM	20	.0 0 0
	VERN	21	.0 0 0

DEADBAND			
ATT	NORM	22	3.0 0
	VERN	23	0.0 3 3

RATE	NORM	24	0.2 0
	VERN	25	.0 2 0

JET OPT	P	26	3
	Y	27	3

CNTL ACCEL 28 0

CAP/26/0/BAS

CAP/26/0/BAS

B6

TOP

(reduced copy)

TOP

B5

DAP B6

PURPOSE:

RENDEZVOUS

B6

TRANSLATION
PULSE

15 0.0 5

ROTATION

DSC RT	NORM	16	0.5	0	0
	VERN	17	0.2	0	0
PULSE	NORM	18	0.0	4	
	VERN	19	0.0	0	2
COMP	NORM	20	.0	0	
	VERN	21	.0	0	0

DEADBAND

ATT	NORM	22	2.0	0	
	VERN	23	1.0	0	0

RATE	NORM	24	0.2	0	
	VERN	25	.0	2	0

JET OPT	P	26	1		
	Y	27	1		

CNTL ACCEL 28 0

CAP/26/0/BAS

CAP/26/0/BAS

B7

TOP

TOP

B6

DAP B7

PURPOSE:

PROX OPS

B7

TRANSLATION
PULSE

15 0.0 1

ROTATION

DSC RT	NORM	16	0.2	0	0
	VERN	17	0.2	0	0
PULSE	NORM	18	0.0	4	
	VERN	19	0.0	0	2
COMP	NORM	20	.0	0	
	VERN	21	.0	0	0

DEADBAND

ATT	NORM	22	2.0	0	
	VERN	23	1.0	0	0

RATE	NORM	24	0.2	0	
	VERN	25	.0	2	0

JET OPT	P	26	1		
	Y	27	1		

CNTL ACCEL 28 0

CAP/26/0/BAS

CAP/26/0/BAS

B8

TOP

(reduced copy)

7-14

CAP/26/FIN C

TOP

B7

DAP B8

PURPOSE: (Blank)

B8

TRANSLATION
PULSE 15

ROTATION
DSC RT NORM 16
VERN 17
PULSE NORM 18
VERN 19
COMP NORM 20
VERN 21

DEADBAND
ATT NORM 22
VERN 23

RATE NORM 24
VERN 25

JET OPT P 26
Y 27

CNTL ACCEL 28

CAP/26/0/BAS

CAP/26/0/BAS

TOP

B8

CAP/26/0/BAS

TOP

(reduced copy)

TOP
BACK COVER 'DAP BOOK'

HOOK
VELCRO

CAP/26/0/BAS

ATTITUDE
TIMELINE

ATTITUDE
TIMELINE

Attitude Timeline

MET	MNVR OPTION	DAP	E/S	REMARKS	EVENT
1 0/01:10:00 19:56	TRK TGT=2 BV=3(90.00, .00) OM=261.00	A1 DSC .200 AUTO DB AT 5.00 NORM DB RT .20	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 81	-ZLV NOSE SOUTH PLBD OPENING
2 0/04:19:00 32:48	MNVR R+ 8.00 P+267.00 Y+358.00	B1 DSC .200 AUTO DB AT 1.00 VERN DB RT .02	SUN T 91 P 183		TDRS DIRECT CHECK ATT
3 0/05:33:00 34:54	TRK TGT=4 BV=5(261.50, .00) OM=270.00 DEPLOY 0/06:13:33	A11 DSC .200 AUTO DB AT 1.00 VERN DB RT .01	SUN T 99 P 180	INRTL R+ 15 P+279 Y+339	TDRS DEPLOY ATTITUDE
4 0/06:14:33	ROT BV=5(.00,270.00)	A11 DSC .430 MAN DB AT 1.00 NORM DB RT .20		ROT R+ 15 START P+279 ATT Y+339	-2.2 FPS SEP AND BEGIN -70 DEGREE PITCH
5 0/06:14:41	ROT BV=5(.00,270.00)	A11 DSC .860 MAN DB AT 1.00 NORM DB RT .20		ROT R+ 14 START P+275 ATT Y+338	SEP CUTOFF CONTINUE -70 DEGREE PITCH
6 0/06:15:59	CANCEL	A11 DSC .200 MAN DB AT 1.00 NORM DB RT .20	SUN T 28 P 180	INRTL R+344 P+203 Y+340	END OF -70 DEGREE PITCH
7 0/06:20:33 25:16	MNVR R+191.00 P+ 13.00 Y+338.00	B1 DSC .500 AUTO DB AT 3.00 NORM DB RT .20	SUN T 143 P 109	TIG 0/06:28:33	OMS SEP ATTITUDE
8 0/06:28:59 32:05	TRK TGT=1 BV=3(90.00, .00) OM= .00	B1 DSC .500 AUTO DB AT 3.00 NORM DB RT .20	EARTH T 90 P 0		IUS VIEWING ATTITUDE
9 0/06:54:33 07:00:22	TRK TGT=1 BV=5(310.00, .00) OM= 22.73	B1 DSC .500 AUTO DB AT 3.00 NORM DB RT .20	SUN T 71 P 145		WINDOW PROTECT ATTITUDE
10 0/07:13:33	CANCEL	B1 DSC .200 MAN DB AT 1.00 VERN DB RT .02	SUN T 71 P 145	INRTL R+102 P+197 Y+ 58	INRTL HOLD
11 0/08:18:00 21:44	MNVR R+ 50.00 P+283.00 Y+ 76.00	A1 DSC .200 AUTO DB AT 1.00 VERN DB RT .02	SUN T 105 P 115	STAR PR A1	IMU ALIGN
12 0/08:40:00 51:47	TRK TGT=2 BV=3(90.00, .00) OM=180.00	A1 DSC .200 AUTO DB AT 1.00 VERN DB RT .02	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 0	-ZLV NOSE FWD

Attitude Timeline

MET	MNVR OPTION		DAP	E/S	REMARKS	EVENT
1 0/20:24:00 36:06	MNVR R+ 50.00 P+283.00 Y+ 76.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 105 P 115	STAR PR A1 -Z COAS:16 -Y ST:37	IMU ALIGN COAS CAL
2 0/20:50:00 21:01:48	TRK TGT=2 BV=3(90.00, .00) OM=180.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 0	-ZLV NOSE FWD
3 0/23:45:00 59:11	MNVR R+ 62.00 P+207.00 Y+ 24.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 47 P 163		KU-BAND ATT
4 1/00:55:00 01:09:23	TRK TGT=2 BV=3(90.00, .00) OM=180.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 0	-ZLV NOSE FWD
5 1/06:59:00 07:11:05	MNVR R+ 50.00 P+283.00 Y+ 76.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 106 P 115	STAR PR A1	IMU ALIGN
6 1/07:20:00 31:32	TRK TGT=2 BV=3(90.00, .00) OM=180.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 0	-ZLV NOSE FWD
7 1/19:05:00 17:01	MNVR R+ 50.00 P+283.00 Y+ 76.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 106 P 115	STAR PR A1 +X COAS:23	IMU ALIGN COAS CAL
8 1/20:16:00 26:32	TRK TGT=2 BV=5(17.40, .00) OM=160.40	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 17 P 359	LVLH R+161 P+286 Y+354	ELRAD SUNSET
9 1/22:21:00 32:46	TRK TGT=2 BV=5(17.30, .00) OM= 19.30	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 17 P 2	LVLH R+ 18 P+254 Y+354	ELRAD SUNRISE
10 2/00:00:00 07:07	MNVR R+ 62.00 P+207.00 Y+ 24.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 48 P 163		KU-BAND ATT
11 2/01:10:00 24:34	TRK TGT=2 BV=3(90.00, .00) OM=180.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 0	-ZLV NOSE FWD
12 2/05:40:00 51:59	MNVR R+ 50.00 P+283.00 Y+ 76.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 107 P 115		IMU ALIGN

Attitude Timeline

MET	MNVR OPTION		DAP	E/S	REMARKS	EVENT
1 2/06:00:00 11:17	TRK TGT=2 BV=3(90.00, .00) OM=180.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 0	-ZLV NOSE FWD
2 2/17:46:00 57:55	MNVR R+ 50.00 P+283.00 Y+ 76.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 107 P 115	STAR PR A1	IMU ALIGN
3 2/18:05:00 16:09	TRK TGT=2 BV=3(90.00, .00) OM=180.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 0	-ZLV NOSE FWD
4 2/22:25:00 39:46	MNVR R+ 62.00 P+207.00 Y+ 24.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 49 P 164		KU-BAND ATT
5 3/00:05:00 19:46	TRK TGT=2 BV=3(90.00, .00) OM=180.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 0	-ZLV NOSE FWD
6 3/05:52:00 06:03:50	MNVR R+ 50.00 P+283.00 Y+ 76.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 108 P 115	STAR PR A1	IMU ALIGN
7 3/06:10:00 21:01	TRK TGT=2 BV=3(90.00, .00) OM=180.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	EARTH T 90 P 0	LVLH R+180 P+ 0 Y+ 0	-ZLV NOSE FWD
8 3/17:58:00 18:09:44	MNVR R+ 50.00 P+283.00 Y+ 76.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 108 P 115	STAR PR A1	IMU ALIGN
9 3/18:20:00 26:43	TRK TGT=4 BV=5(184.00, .00) OM=270.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 176 P 179	INRTL R+ 30 P+350 Y+ 12	TAIL SUN COLD SOAK
10 3/21:00:00 05:52	MNVR R+344.00 P+310.00 Y+ 25.00	A1 AUTO VERN	DSC .200 DB AT 1.00 DB RT .02	SUN T 138 P 190	-Z ST:26 -Y ST:14	D/O IMU ALIGN
11 3/21:20:00 23:38	MNVR R+ 90.00 P+252.00 Y+356.00	B1 AUTO NORM	DSC .500 DB AT 3.00 DB RT .20	SUN T 78 P 104	-Z ST:23 -Y ST:37	IMU VERIF
12 3/21:30:00 32:29	MNVR R+ 61.00 P+179.00 Y+332.00	B1 AUTO NORM	DSC .500 DB AT 3.00 DB RT .20	SUN T 15 P 49	TIG 3/23:54:30	DEORBIT ATT

8-4

CAP/26/FIN C

ALL ATTITUDES IN MEAN OF 1950

